
**Space data and information transfer
systems — Packet telemetry**

*Systèmes de transfert des informations et données spatiales —
Télémétrie par paquets*



Foreword

ISO (the International Organization for Standardization) is a worldwide federation of national standards bodies (ISO member bodies). The work of preparing International Standards is normally carried out through ISO technical committees. Each member body interested in a subject for which a technical committee has been established has the right to be represented on that committee. International organizations, governmental and non-governmental, in liaison with ISO, also take part in the work. ISO collaborates closely with the International Electrotechnical Commission (IEC) on all matters of electrotechnical standardization.

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Space data and information transfer systems — Packet telemetry

1 Scope

This International Standard specifies the requirements for spacecraft packet telemetry systems.

2 Requirements

Requirements are the technical recommendations made in the following publication (reproduced on the following pages), which is adopted as an International Standard:

CCSDS 102.0-B-4, November 1995, *Recommendation for space data system standards — Packet telemetry*.

For the purposes of international standardization, the modifications outlined below shall apply to the following pages of publication CCSDS 102.0-B-4.

Pages i to vi

This part contains information which is relevant to the CCSDS publication only.

Page 1-4

Add the following information to the references indicated in chapter 1.7:

- [2] Document CCSDS 101.0-B-3, May 1992, is equivalent to ISO 11754:1994.
- [3] Document CCSDS 301.0-B-2, April 1990, is equivalent to ISO 11104:1991.
- [4] Document CCSDS 202.0-B-2, November 1992, is equivalent to ISO 12172:—1).
- [6] Document CCSDS 701.0-B-2, November 1992, is equivalent to ISO 13420:—1).

3 Revision of publication CCSDS 102.0-B-4

It has been agreed with the Consultative Committee for Space Data Systems that ISO/TC 20/SC 13 will be consulted in the event of any revision or amendment of publication CCSDS 102.0-B-4. To this end, NASA will act as a liaison body between CCSDS and ISO.

1) To be published.

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Consultative Committee for Space Data Systems

RECOMMENDATION FOR SPACE
DATA SYSTEM STANDARDS

PACKET TELEMETRY

CCSDS 102.0-B-4

BLUE BOOK

November 1995



DEDICATION

This document is dedicated to the memory of Mr. Daniel Rouat of the European Space Agency. He was a major contributor to the recommendations for Telecommand and Packet Telemetry, including the revisions in this issue. His vision and technical contributions inspired many people throughout the world, and he will be deeply missed by his many friends and colleagues in the CCSDS.

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CCSDS RECOMMENDATION FOR PACKET TELEMETRY

AUTHORITY

Issue:	Blue Book, Issue 4
Date:	November 1995
Location:	Toulouse, France

This document has been approved for publication by the Management Council of the Consultative Committee for Space Data Systems (CCSDS) and represents the consensus technical agreement of the participating CCSDS Member Agencies. The procedure for review and authorization of CCSDS Recommendations is detailed in reference [1], and the record of Agency participation in the authorization of this document can be obtained from the CCSDS Secretariat at the address below.

This document is published and maintained by:

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National Aeronautics and Space Administration
Washington, DC 20546, USA

CCSDS RECOMMENDATION FOR PACKET TELEMETRY

STATEMENT OF INTENT

The **CONSULTATIVE COMMITTEE FOR SPACE DATA SYSTEMS (CCSDS)** is an organisation officially established by the management of member space Agencies. The Committee meets periodically to address data systems problems that are common to all participants, and to formulate sound technical solutions to these problems. Inasmuch as participation in the CCSDS is completely voluntary, the results of Committee actions are termed **RECOMMENDATIONS** and are not considered binding on any Agency.

This **RECOMMENDATION** is issued by, and represents the consensus of, the CCSDS Plenary body. Agency endorsement of this **RECOMMENDATION** is entirely voluntary. Endorsement, however, indicates the following understandings:

- Whenever an Agency establishes a CCSDS-related **STANDARD**, this **STANDARD** will be in accord with the relevant **RECOMMENDATION**. Establishing such a **STANDARD** does not preclude other provisions which an Agency may develop.
- Whenever an Agency establishes a CCSDS-related **STANDARD**, the Agency will provide other CCSDS member Agencies with the following information:
 - the **STANDARD** itself.
 - the anticipated date of initial operational capability.
 - the anticipated duration of operational service.
- Specific service arrangements shall be made via memoranda of agreement. Neither this **RECOMMENDATION** nor any ensuing **STANDARD** is a substitute for a memorandum of agreement.

No later than five years from its date of issuance, this **RECOMMENDATION** will be reviewed by the CCSDS to determine whether it should: (1) remain in effect without change; (2) be changed to reflect the impact of new technologies, new requirements, or new directions; or (3) be retired or cancelled.

In those instances when a new version of a **RECOMMENDATION** is issued, existing CCSDS-related Agency standards and implementations are not negated or deemed to be non-CCSDS compatible. It is the responsibility of each Agency to determine when such standards or implementations are to be modified. Each Agency is, however, strongly encouraged to direct planning for its new standards and implementations towards the later version of the **RECOMMENDATION**.

CCSDS RECOMMENDATION FOR PACKET TELEMETRY

FOREWORD

This document is a technical **RECOMMENDATION** for use in developing packetised telemetry systems and has been prepared by the **CONSULTATIVE COMMITTEE FOR SPACE DATA SYSTEMS** (CCSDS). The Packet Telemetry concept described herein is the baseline concept for spacecraft-to-ground data communication within missions that are cross-supported between Agencies of the CCSDS.

This **RECOMMENDATION** establishes a common framework and provides a common basis for the data structures of spacecraft telemetry streams. It allows implementing organisations within each Agency to proceed coherently with the development of compatible derived Standards for the flight and ground systems that are within their cognizance. Derived Agency Standards may implement only a subset of the optional features allowed by the **RECOMMENDATION** and may incorporate features not addressed by the **RECOMMENDATION**.

Through the process of normal evolution, it is expected that expansion, deletion or modification to this document may occur. This **RECOMMENDATION** is therefore subject to CCSDS document management and change control procedures which are defined in Reference [1].

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- Deutsche Forschungsanstalt für Luft- und Raumfahrt e.V. (DLR)/Germany.
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- European Telecommunications Satellite Organization (EUTELSAT)/Europe.
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- United States Geological Survey (USGS)/USA.

CCSDS RECOMMENDATION FOR PACKET TELEMETRY

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Changes not Compatible with the Previous Issue

The option of Source Packet Segmentation has been eliminated.

Editorial Changes

The definition of Source Packet Grouping has been clarified.

Minor format changes have been made based on the specifications of the CCSDS Publications Manual.

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1 INTRODUCTION

1.1 PURPOSE

The purpose of this document is to establish a common **RECOMMENDATION** for the implementation of spacecraft "Packet Telemetry" systems by the Agencies participating in the **CONSULTATIVE COMMITTEE FOR SPACE DATA SYSTEMS (CCSDS)**.

1.2 SCOPE

PACKET TELEMETRY is a concept which facilitates the transmission of space-acquired data from source to user in a standardised highly automated manner. **PACKET TELEMETRY** provides a mechanism for implementing common data transport structures and protocols which may enhance the development and operation of space mission systems.

This **RECOMMENDATION** addresses the following two processes:

- The end-to-end transport of space mission data sets from source application processes located in space to distributed user application processes located on the ground.
- The intermediate transfer of these data sets through space data acquisition networks, which contain spacecraft, radio links, tracking stations, ground communications circuits and mission control centres as some of their components.

This **RECOMMENDATION** is limited to describing the telemetry formats which are generated by the spacecraft in order to execute its role in the above processes. The services corresponding to these formats are defined in Reference [8]. The CCSDS channel coding and synchronisation mechanisms required to implement space-to-ground data links of acceptable quality are defined in Reference [2].

An overview of the **PACKET TELEMETRY** Concept is given in Chapter 2.

1.3 APPLICABILITY

This **RECOMMENDATION** applies to the creation of Agency standards and to the future exchange of **PACKET TELEMETRY** between CCSDS Agencies in cross-support situations. The **RECOMMENDATION** includes comprehensive specification of the structure of data streams that are generated by remote space vehicles for telemetering to space mission data processing facilities (which are usually located on Earth). The **RECOMMENDATION** does not attempt to define the architecture or configuration of these data processing facilities, except to describe assumed ground data handling services which affect the selection of certain on-board formatting options.

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The **RECOMMENDATION** specifies a wide range of formatting capabilities which may facilitate a high degree of flexibility in the design of spacecraft data acquisition systems; however, compatibility with the **PACKET TELEMETRY** concept may be realised by only implementing a narrow subset of these capabilities. Some “Application Notes” which discuss how different levels of compatibility may be achieved are included in Reference [5].

1.4 RATIONALE

The CCSDS believes it is important to document the rationale underlying the recommendations chosen, so that future evaluations of proposed changes or improvements will not lose sight of previous decisions. The concept and rationale for **PACKET TELEMETRY** may also be found in Reference [5].

1.5 STRUCTURE OF THE DOCUMENT

For the designation of text partitions the following conventions will be used:

- text designated by one number belongs to a Chapter;
- text designated by two numbers belongs to a Section;
- text designated by three numbers belongs to a Sub-Section;
- text designated by four numbers belongs to a Paragraph;
- text designated by a lower case letter belongs to an Item.

All specifications are contained in Chapters 3 and 5 of this **RECOMMENDATION**. They are identified by an Item Number consisting of the number of the text partition as defined above, and a lower case letter. The conventions and definitions applied in these specifications are itemised in Section 1.6.

All other text and all figures in these chapters represent comments to these specifications. All comments are printed in italics.

The contents of the specifications take precedence over those of the comments.

All terms printed in bold-face upper-case are referenced in the Index.

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1.6 CONVENTIONS AND DEFINITIONS

The following Items contain the conventions which have been used throughout this **RECOMMENDATION**.

- (a) To identify each bit in an **N-BIT FIELD** the first bit in the field to be transferred (i.e., the most left justified when drawing a figure) is defined to be "Bit 0"; the following bit is defined to be "Bit 1" and so on up to "Bit

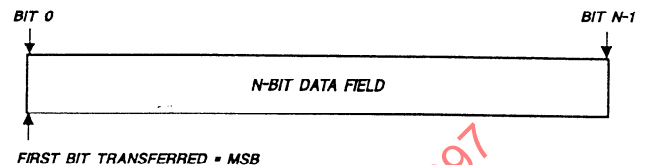


Figure 1-1: Bit Numbering Convention

- N-1". When the field is used to express a binary value (such as a counter), the **MOST SIGNIFICANT BIT** shall be the first bit of the field, i.e., "Bit 0" (see Figure 1-1).
- (b) In accordance with modern data communication practice, spacecraft data fields are often grouped into 8-bit words which conform to convention 1.6.a. Throughout this **RECOMMENDATION**, such an 8-bit word is termed **OCTET**.
- (c) The numbering for **OCTETs** within a data structure starts with 0.
- (d) The term **MISSION PHASE** designates a period of a mission during which specified telemetry characteristics are fixed. The transition between two consecutive **MISSION PHASEs** may cause an interruption of the telemetry services.
- (e) Certain characteristics of the data structures specified in this **RECOMMENDATION** are required to remain unchanged throughout a **MISSION PHASE** or throughout all **MISSION PHASEs**. In these cases the term "static" is used to specify characteristics which remain unchanged either with respect to an **APPLICATION PROCESS IDENTIFIER** (for definition see Paragraph 3.1.2.3), or within a specific **VIRTUAL CHANNEL** (for definition see Item 5.e) or within a specific **MASTER CHANNEL** (for definition see Item 5.d).
- (f) **IDLE DATA** is data which carries no information, but is sent to meet timing or synchronisation requirements. The bit pattern of **IDLE DATA** is not specified.

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1.7 REFERENCES

The following documents are referenced in the text of this Recommendation. At the time of publication, the editions indicated were valid. All documents are subject to revision, and users of this Recommendation are encouraged to investigate the possibility of applying the most recent editions of the documents indicated below. The CCSDS Secretariat maintains a register of currently valid CCSDS Recommendations.

- [1] *Procedures Manual for the Consultative Committee for Space Data Systems*. CCSDS A00.0-Y-6. Yellow Book. Issue 6. Washington, D.C.: CCSDS, May 1994.
- [2] *Telemetry Channel Coding*. Recommendation for Space Data Systems Standards, CCSDS 101.0-B-3. Blue Book. Issue 3. Washington, D.C.: CCSDS, May 1992.
- [3] *Time Code Formats*. Recommendation for Space Data Systems Standards, CCSDS 301.0-B-2. Blue Book. Issue 2. Washington, D.C.: CCSDS, April 1990.
- [4] *Telecommand Part 2—Data Routing Service*. Recommendation for Space Data Systems Standards, CCSDS 202.0-B-2. Blue Book. Issue 2. Washington, D.C.: CCSDS, November 1992.
- [5] *Telemetry Summary of Concept and Rationale*. Report Concerning Space Data Systems Standards, CCSDS 100.0-G-1. Green Book. Issue 1. Washington, D.C.: CCSDS, December 1987.
- [6] *Advanced Orbiting Systems, Networks and Data Links: Architectural Specification*. Recommendation for Space Data Systems Standards, CCSDS 701.0-B-2. Blue Book. Issue 2. Washington, D.C.: CCSDS, November 1992.
- [7] *CCSDS Global Spacecraft Identification Field Code Assignment Control Procedures*. Recommendation for Space Data Systems Standards, CCSDS 320.0-B-1. Blue Book. Issue 1. Washington, D.C.: CCSDS, October 1993.
- [8] *Packet Telemetry Services*. Recommendation for Space Data Systems Standards, CCSDS 103.0-B-1. Blue Book. Issue 1. Washington, D.C.: CCSDS, May 1996.

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2 OVERVIEW

This **PACKET TELEMETRY RECOMMENDATION** describes data structures used to transport data from data sources on board a space vehicle to data sinks on the ground, as shown in Figure 2-1.



Figure 2-1: CCSDS Packet Telemetry Data System

2.1 THE PACKET TELEMETRY CONCEPT

The essence of the packet telemetry concept is to permit multiple application processes running in on-board sources to create units of data as best suits each data source, and then to permit the on-board data system to transmit these data units over a space-to-ground communications channel in a way that enables the ground system to recover the individual data units with high reliability and provide them to the data sinks in sequence. These on-board sources are either instruments or sub-systems.

To accomplish these functions, this Recommendation defines two data structures — **SOURCE PACKETS** and **TRANSFER FRAMES** — and a multiplexing process to interleave **SOURCE PACKETS** from various **APPLICATION PROCESSES** into **TRANSFER FRAMES**.

2.2 SOURCE PACKET

The **SOURCE PACKET**, which in the following text may also be termed packet, is a data structure generated by an on-board **APPLICATION PROCESS** in a way that is responsive to the needs of that process. It can be generated at fixed or variable intervals and may be fixed or variable in length. Aside from a packet header that identifies the source and characteristics of the packet, the internal data content of the **SOURCE PACKET** is completely under the control of the **APPLICATION PROCESS**.

The **SOURCE PACKET** allows each **APPLICATION PROCESS** within a data source to optimise the size and structure of its data set with a minimum of constraints imposed by the spacecraft-to-ground transport system. Each data source is thus able to define its data organisation independently of other data sources and to adapt this organisation to the various modes of the instrument or sub-system.

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The **SOURCE PACKET PRIMARY HEADER** contains an **APPLICATION PROCESS IDENTIFIER** used to route the packet to its destination sink. The header also carries information about the length, sequence, and other characteristics of the packet. An optional **SOURCE PACKET SECONDARY HEADER** is provided for standardised time-tagging of **SOURCE PACKETS**, and to carry application-unique ancillary data.

2.3 TRANSFER FRAME

The **TRANSFER FRAME** is a data structure that provides an envelope for transmitting packetised data over a noisy space-to-ground channel. It carries information in the **TRANSFER FRAME PRIMARY HEADER** that permits the ground system to route the **TRANSFER FRAMES** to their intended destination. The **TRANSFER FRAME** is of fixed length (for a given **PHYSICAL CHANNEL** during a **MISSION PHASE**). It is compatible with the CCSDS Recommendation for Telemetry Channel Coding (including synchronisation) (see Reference [2]); thus the transmitted data can be recovered with extremely high reliability.

Multiple, individual, asynchronous **APPLICATION PROCESSES** on board a space vehicle can generate variable-length **SOURCE PACKETS** at different rates, and these **SOURCE PACKETS** can then be multiplexed together into a synchronous stream of fixed-length coded **TRANSFER FRAMES** for reliable transmission to the ground.

The **TRANSFER FRAME PRIMARY HEADER** provides the necessary elements to allow the variable-length **SOURCE PACKETS** from a number of **APPLICATION PROCESSES** on a spacecraft to be multiplexed into a sequence of fixed-length frames. Short packets may be contained in a single frame, while longer ones may span two or more frames. Since a packet can begin or end at any place in a frame, the entire data field of every frame can be used to carry data; there is no need to tune the sizes of packets or their order of occurrence to fit the frames.

The mechanism of **IDLE PACKETS** is provided for the case where a frame must be released and insufficient packet data is available. Further, frames containing **IDLE DATA** in the **TRANSFER FRAME DATA FIELD** are defined to keep the data capture element in synchronisation in the absence of data. (Other fields in the frame may still contain valid data.)

On the ground, the information in the frame and packet headers allows the data acquisition system to extract packets in a standardised way.

In addition to packets, the **TRANSFER FRAME** can carry two optional fields, the **TRANSFER FRAME SECONDARY HEADER** and the **OPERATIONAL CONTROL FIELD**. The **TRANSFER FRAME SECONDARY HEADER** can be used to carry fixed-length mission-specific data. The **OPERATIONAL CONTROL FIELD** can be used to provide the status of telecommand or other spacecraft operations activities. Instead of packets the **TRANSFER FRAME** can carry **PRIVATELY DEFINED DATA**.

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2.4 SHARING TRANSMISSION RESOURCES

As most space communication systems are capacity-limited, multiple users must share access to the downlink data channel, so the on-board data system must be able to manage the data flow to the ground in an orderly manner. In addition, different types of data may be handled differently on the spacecraft or on the ground. This Recommendation provides the method of **VIRTUAL CHANNELISATION** for controlling the data flow.

VIRTUAL CHANNELISATION is the mechanism that allows the various sources which generate packets to be “virtually” given exclusive access to a **PHYSICAL CHANNEL** by assigning them transmission capacity on a frame-by-frame basis. Each **TRANSFER FRAME** is identified as belonging to one of the up to eight **VIRTUAL CHANNELS**. **VIRTUAL CHANNELISATION** is normally used to separate sources or destinations with different characteristics. For example, if a payload contains an imaging instrument which produces packets containing many thousands of octets, and a number of other instruments which generate smaller packets, a possible system architecture would be to assign the imaging instrument packets to one **VIRTUAL CHANNEL** and to handle the rest by multiplexing them onto a second **VIRTUAL CHANNEL**. **VIRTUAL CHANNELS** may also be used to separate real-time packets from recorded packets, both on the spacecraft and on the ground, and to allow easy separation on the ground of data streams that are to be sent to different destinations.

Figure 2-2 shows the flow of telemetry data from several on-board packet sources (instruments or sub-systems), through to the delivery of the same data to **SINK PROCESSES** on the ground. At the top of the figure, generation of **SOURCE PACKETS** from **APPLICATION PROCESSES** in several data sources is shown. These packets are multiplexed into the **TRANSFER FRAMES** of several **VIRTUAL CHANNELS**. These **TRANSFER FRAMES** are transmitted to the ground, using appropriate error protection and synchronisation techniques. On the ground they are demultiplexed into **VIRTUAL CHANNELS**, and the packets are extracted. **SOURCE PACKETS** are then delivered to **SINK PROCESSES**, shown at the bottom of the figure, using the **APPLICATION PROCESS IDENTIFIERS** in the **SOURCE PACKET** headers for routing. **SOURCE PACKETS** with a given **APPLICATION PROCESS IDENTIFIER** may be delivered to one or more **SINK PROCESSES**. Packets may be time-ordered prior to delivery using the information in the **PACKET PRIMARY HEADER** and the **PACKET SECONDARY HEADER**.

2.5 APPLICATION NOTES

Application Notes, which describe how compatibility with these various data structures may be achieved, are presented in Reference [5], along with key elements of the rationale behind **PACKET TELEMETRY**.

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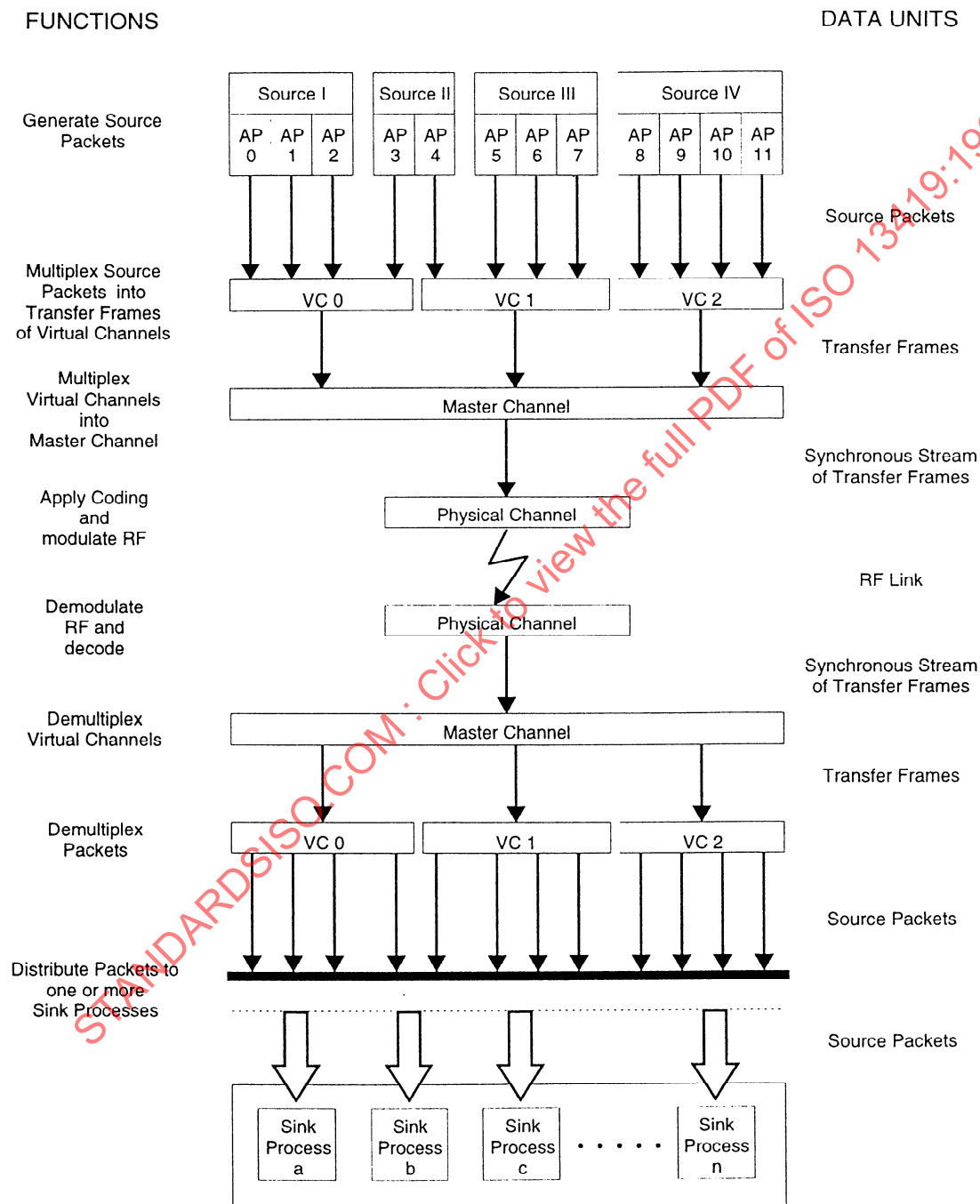


Figure 2-2: Example of Telemetry Data Flow

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3 SOURCE PACKET

- a. A **SOURCE PACKET**, which in the following text may also be termed **PACKET**, shall encapsulate a block of observational and ancillary application data which is to be transmitted from an **APPLICATION PROCESS** in space to one or several **SINK PROCESSES** on the ground.

- b. The **SOURCE PACKET** shall consist of two major fields, positioned contiguously, in the following sequence:

	Length in bits
— PACKET PRIMARY HEADER (mandatory)	48
— PACKET DATA FIELD (mandatory)	variable

- c. The **SOURCE PACKET** shall consist of at least 7 and at most 65542 octets.

- d. A **SOURCE PACKET** which contains **IDLE DATA** in its **PACKET DATA FIELD** is called an **IDLE PACKET**.

Idle Packets may be generated by the on-board data system when needed to maintain synchronisation of the data transport and the packet extraction processes.

- e. A series of **SOURCE PACKETS** generated consecutively by a single **APPLICATION PROCESS** may be designated as a **GROUP OF SOURCE PACKETS**.

Figure 3-1 shows the format of the Source Packet as specified above including the sub-formats to be specified in the following sections.

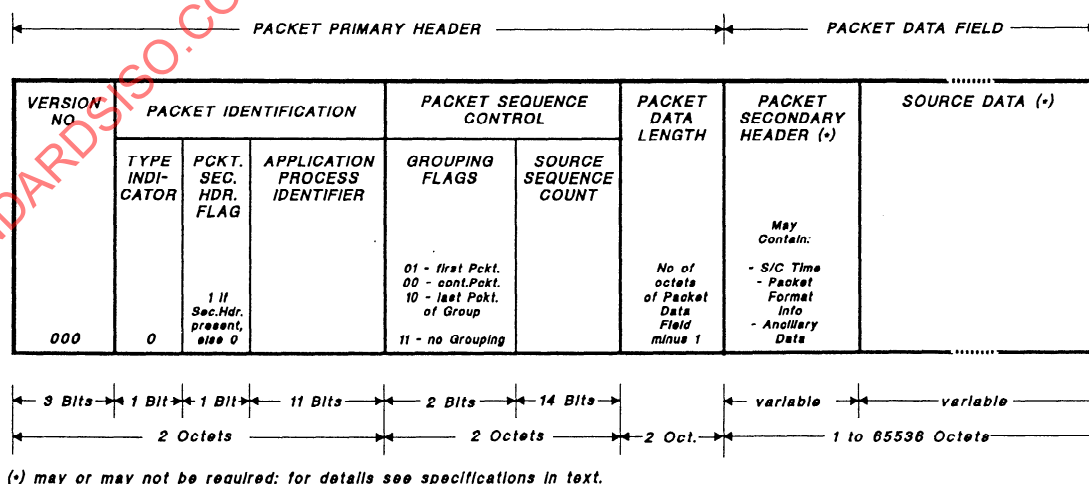


Figure 3-1: Source Packet Format

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- ¹ Version Number 100 was specified in previous issues of this document for the Source Packet Segment, which is no longer defined.

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3.1.2.1 TYPE INDICATOR

- a. Bit 3 of the **PACKET PRIMARY HEADER** shall contain the **TYPE INDICATOR** indicating the type of data unit.
- b. The **TYPE INDICATOR** shall be set to "0".

Because CCSDS telecommand uses a similar packet structure, the type indicator distinguishes between telemetry and telecommand data units (for telecommand packets the type indicator will be set to "1"; see Reference [4]).

3.1.2.2 PACKET SECONDARY HEADER FLAG

- a. Bit 4 of the **PACKET PRIMARY HEADER** shall contain the **PACKET SECONDARY HEADER FLAG**.
- b. The **PACKET SECONDARY HEADER FLAG** shall indicate the presence or absence of the **PACKET SECONDARY HEADER** within this **SOURCE PACKET**. It shall be "1", if a **PACKET SECONDARY HEADER** is present; it shall be "0", if a **PACKET SECONDARY HEADER** is not present.
- c. The **PACKET SECONDARY HEADER FLAG** shall be static with respect to the **APPLICATION PROCESS IDENTIFIER** throughout a **MISSION PHASE**.
- d. The **PACKET SECONDARY HEADER FLAG** shall be set to "0" for **IDLE PACKETS**.

3.1.2.3 APPLICATION PROCESS IDENTIFIER

- a. Bits 5–15 of the **PACKET PRIMARY HEADER** shall contain the **APPLICATION PROCESS IDENTIFIER**.
- b. The **APPLICATION PROCESS IDENTIFIER** shall be different for different **APPLICATION PROCESSES** on the same **MASTER CHANNEL** (for the definition of the **MASTER CHANNEL** see Item 5.d).
- c. For **IDLE PACKETS** the **APPLICATION PROCESS IDENTIFIER** shall be "1111111111", i.e., "all ones".

This identifier is tailored to local mission needs and is therefore assigned by mission management. Users should note that ground data accounting considerations may limit the number of different Application Processes which may be active simultaneously. Certain Application Process Identifiers have been reserved in Reference [6] to be used for specific purposes.

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3.1.3 PACKET SEQUENCE CONTROL FIELD

- a. The **PACKET SEQUENCE CONTROL FIELD** shall be contained within bits 16–31 of the **PACKET PRIMARY HEADER**.
- b. This 16-bit field shall be sub-divided into two sub-fields as follows:

—	GROUPING FLAGS	Length in bits 2
—	SOURCE SEQUENCE COUNT	14

The Packet Sequence Control Field provides a sequential count of the packets generated with the same Application Process Identifier, and if the grouping feature is applied, provides information on the position of a Source Packet in a group.

3.1.3.1 GROUPING FLAGS

- a. Bits 16 and 17 of the **PACKET PRIMARY HEADER** shall contain the **GROUPING FLAGS**.
- b. The **GROUPING FLAGS** shall be set as follows:
 - “01” for the first **SOURCE PACKET** of a group;
 - “00” for a continuing **SOURCE PACKET** of a group;
 - “10” for a last **SOURCE PACKET** of a group.
- c. For a **SOURCE PACKET** not belonging to a **GROUP OF SOURCE PACKETS** the **GROUPING FLAGS** shall be set to “11”.
- d. All **SOURCE PACKETS** belonging to a specific **GROUP OF SOURCE PACKETS** shall originate from the same **APPLICATION PROCESS** identified by a unique **APPLICATION PROCESS IDENTIFIER**.

The use of a Group of Source Packets is outside the scope of this Recommendation.

3.1.3.2 SOURCE SEQUENCE COUNT

- a. Bits 18–31 of the **PACKET PRIMARY HEADER** shall contain the **SOURCE SEQUENCE COUNT**.
- b. The **SOURCE SEQUENCE COUNT** shall provide the sequential binary count of each **SOURCE PACKET** generated by an **APPLICATION PROCESS** identified by a unique **APPLICATION PROCESS IDENTIFIER**.
- c. The **SOURCE SEQUENCE COUNT** shall be continuous, modulo 16384.

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- d. **IDLE PACKETs** are not required to increment the **SOURCE SEQUENCE COUNT**.
- e. A re-setting of the **SOURCE SEQUENCE COUNT** before reaching 16383 shall not take place unless it is unavoidable.

The purpose of the field is to order this packet with other packets generated by the same Application Process, even though their natural order may have been disturbed during transport to the user's processor on the ground.

The field will normally be used in conjunction with a Time Code (see Paragraph 3.2.1.1; its insertion is, however, not mandatory) to provide unambiguous ordering; it is therefore essential that the resolution of the time code is sufficient for this code to increment at least once between successive recyclings of the Source Sequence Count.

If the Source Sequence Count is re-set due to an unavoidable re-initialisation of a process the completeness of a sequence of Source Packets cannot be determined.

3.1.4 PACKET DATA LENGTH FIELD

- a. The **PACKET DATA LENGTH FIELD** shall be contained within bits 32–47 of the **PACKET PRIMARY HEADER**.
- b. This 16-bit field shall contain a binary number equal to the number of octets in the **PACKET DATA FIELD** minus 1.
- c. The value contained in the **PACKET DATA LENGTH FIELD** may be variable and shall be in the range of 0 to 65535, corresponding to 1 to 65536 octets.

Users should recognise that although very long packets are permissible, these may present special problems in terms of data link monopolisation, source data buffering, and network accountability during transfer across the unique channel from the spacecraft to the ground and may add complexity to ground processing. The Recommendation therefore provides the means to assign these packets to individual Virtual Channels (see Chapter 5). An additional measure could be to limit the maximum length of the Source Packets for a specific mission or mission phase.

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3.2 PACKET DATA FIELD

- a. The **PACKET DATA FIELD** shall follow, without gap, the **PACKET PRIMARY HEADER**.
- b. The **PACKET DATA FIELD** is mandatory and shall consist of at least one of the two fields, positioned contiguously, in the following sequence:

— PACKET SECONDARY HEADER	Length in bits variable
— SOURCE DATA FIELD	variable
- c. The **PACKET DATA FIELD** shall contain at least one octet.

3.2.1 PACKET SECONDARY HEADER

- a. If present, the **PACKET SECONDARY HEADER** shall follow, without gap, the **PACKET DATA LENGTH FIELD**.
- b. The **PACKET SECONDARY HEADER** is mandatory if no **SOURCE DATA FIELD** is present; otherwise it is optional. The presence or absence of a **PACKET SECONDARY HEADER** shall be signalled by the **PACKET SECONDARY HEADER FLAG** within the **PACKET IDENTIFICATION FIELD** (see Paragraph 3.1.2.2).
- c. If present, the **PACKET SECONDARY HEADER** shall consist of either
 - a **PACKET SECONDARY HEADER DATA FIELD**;
 - or a **PACKET SECONDARY HEADER TIME CODE FIELD**;
 - or a **PACKET SECONDARY HEADER TIME CODE FIELD** followed by a **PACKET SECONDARY HEADER DATA FIELD**.

The chosen option shall remain static for a specific **APPLICATION PROCESS IDENTIFIER** throughout all **MISSION PHASES**.

The purpose of the Secondary Header is to allow (but not require) a CCSDS-defined means for placing ancillary data (time, internal data field format, spacecraft position/attitude, etc.) within a Source Packet.

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3.2.1.1 PACKET SECONDARY HEADER TIME CODE FIELD

- a. If present, the **PACKET SECONDARY HEADER TIME CODE FIELD** shall consist of an integral number of octets.
- b. The **PACKET SECONDARY HEADER TIME CODE FIELD** shall consist of one of the CCSDS segmented binary or unsegmented binary time codes specified in Reference [3].

The time codes defined in Reference [3] consist of an optional P-Field (Preamble Field), which identifies the time code and its characteristics, and a mandatory T-Field (Time Field). Examples of time codes are CCSDS Unsegmented Time Code and CCSDS Day Segmented Time Code. Examples of characteristics are ambiguity period, epoch, length, and resolution.

- c. The time code selected shall be static for a given **APPLICATION PROCESS IDENTIFIER** throughout all **MISSION PHASES**.
- d. If the characteristics of the time code chosen are allowed to change for an **APPLICATION PROCESS IDENTIFIER**, the **P-FIELD** shall be present.

If the characteristics are static for an Application Process Identifier, the corresponding P-Field need not be present.

- e. The presence or absence of the **P-FIELD** in the **PACKET SECONDARY HEADER TIME CODE FIELD** shall be static for an **APPLICATION PROCESS IDENTIFIER** throughout all **MISSION PHASES**. If present, it shall immediately precede the **T-FIELD**.

For services such as archiving, sorting, processing and correlation with other data sets, the Source Sequence Count may have to be concatenated with a time field in order to unambiguously identify a packet.

See also the comment concerning time code under Paragraph 3.1.3.2.

3.2.1.2 PACKET SECONDARY HEADER DATA FIELD

- a. If present, the **PACKET SECONDARY HEADER DATA FIELD** shall consist of an integral number of octets.

The Data Field may contain any ancillary data necessary for the interpretation of the information contained within the Source Data Field of the Packet. The content and the format of this data are not specified by this Recommendation.

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3.2.2 SOURCE DATA FIELD

- a. If present, the **SOURCE DATA FIELD** shall follow, without gap, either the **PACKET SECONDARY HEADER** (if a **PACKET SECONDARY HEADER** is present) or the **PACKET DATA LENGTH FIELD** (if a **PACKET SECONDARY HEADER** is not present).
- b. The **SOURCE DATA FIELD** is mandatory if no **PACKET SECONDARY HEADER** is present, otherwise it is optional.
- c. The **SOURCE DATA FIELD** shall contain either **SOURCE DATA** from an **APPLICATION PROCESS** or **IDLE DATA**.
- d. The length of the **SOURCE DATA FIELD** may be variable. It shall contain an integral number of octets. See also the specifications in Sub-Section 3.1.4.

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4 [NO LONGER USED]

[This chapter defined the Source Packet Segment, which is no longer defined in this issue.]

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5 **TRANSFER FRAME**

- a. The **TRANSFER FRAME** shall provide the data structure for the transmission of
 - (1) **SOURCE PACKETS**,
 - (2) **IDLE DATA**, and
 - (3) **PRIVATELY DEFINED DATA**

across the downlink channel which connects the spacecraft to a data capture element on the ground.

The start of the Transfer Frame is always signaled by the Attached Sync Marker (ASM), which immediately precedes the Transfer Frame. The ASM is specified in Reference [2]. When the Transfer Frame is synchronously imbedded in a Reed-Solomon codeblock, the ASM signals the start of both.

Privately Defined Data may be specialised high-rate data or other data not suitable for CCSDS Source Packet structuring.

- b. The **TRANSFER FRAME** shall encompass the major fields, positioned contiguously, in the following sequence:

	Length in bits
— TRANSFER FRAME PRIMARY HEADER (mandatory)	48
— TRANSFER FRAME SECONDARY HEADER (optional)	16, 24, ... or 512
— TRANSFER FRAME DATA FIELD (mandatory)	variable
— OPERATIONAL CONTROL FIELD (optional)	32
— FRAME ERROR CONTROL FIELD (mandatory, if Reed-Solomon Encoding is not applied, otherwise optional)	16

- c. The **TRANSFER FRAME** shall be of constant length throughout a specific **MISSION PHASE**. Its length shall be consistent with the specifications contained in Reference [2].

The Telemetry Channel Coding Blue Book (Reference [2]) issued in 1992 limits the maximum Transfer Frame length to 8920 bits.

A change of Frame Length may result in a loss of synchronisation at the data capture element.

- d. All **TRANSFER FRAMES** with the same **TRANSFER FRAME VERSION NUMBER** (see Sub-Section 5.1.1) and the same **SPACECRAFT IDENTIFIER** (see Paragraph 5.1.2.1) on the same **PHYSICAL CHANNEL** constitute a **MASTER CHANNEL**.

In most cases the Master Channel will be identical with the Physical Channel. However, if the Physical Channel also carries Transfer Frames with other Spacecraft Identifiers, a distinction between Master Channel and Physical Channel is necessary; i.e.,

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multiplexing of Transfer Frames with different Spacecraft Identifiers will be performed by the multiplexing of different Master Channels on the same Physical Channel.

- e. A **MASTER CHANNEL** shall consist of between one and eight **VIRTUAL CHANNELS**.

Although Packet Telemetry Systems may be designed to tolerate channel noise, full benefit from Packet Telemetry will require that a high-quality data channel be provided so that packetised data may be adaptively inserted into the frame. The relevant CCSDS Recommendation, Reference [2], describes the coding mechanisms for such a channel, including frame synchronisation and randomisation.

Figure 5-1 illustrates the detailed format of the Transfer Frame.

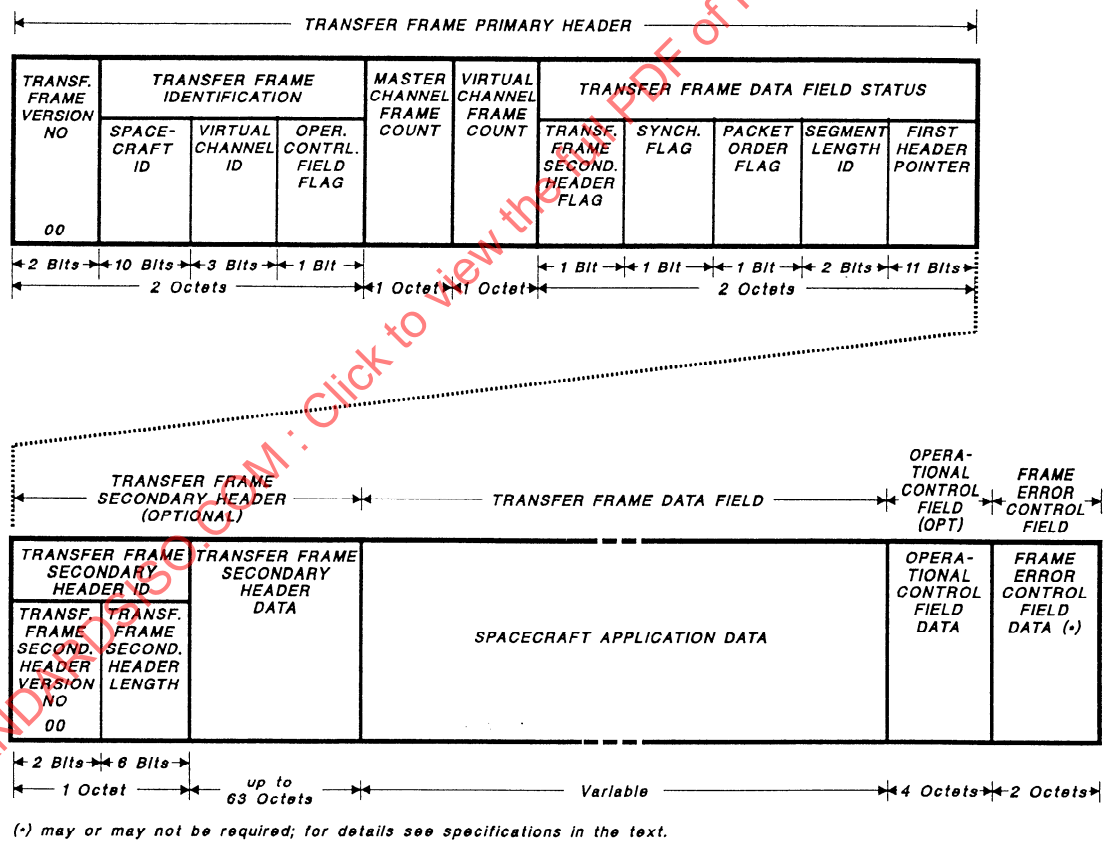


Figure 5-1: Transfer Frame Format

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5.1 TRANSFER FRAME PRIMARY HEADER

- a. The **TRANSFER FRAME PRIMARY HEADER** is mandatory and shall consist of five fields, positioned contiguously, in the following sequence:

	Length in bits
— TRANSFER FRAME VERSION NUMBER	2
— TRANSFER FRAME IDENTIFICATION	14
— MASTER CHANNEL FRAME COUNT	8
— VIRTUAL CHANNEL FRAME COUNT	8
— TRANSFER FRAME DATA FIELD STATUS	16

The Primary Header covers five principal functions:

- *identification of the data unit as a Transfer Frame;*
- *identification of the spacecraft (and possibly of the link, if applicable), which transmitted the telemetered data;*
- *multiplexing of the Virtual Channels into one Master Channel;*
- *providing a counting mechanism for the Virtual Channels and the Master Channel;*
- *providing pointers and other control information so that variable-length Source Packets may be extracted from the Transfer Frame Data Field.*

5.1.1 TRANSFER FRAME VERSION NUMBER

- a. The **TRANSFER FRAME VERSION NUMBER** shall be contained within bits 0–1 of the **TRANSFER FRAME PRIMARY HEADER**.
- b. This 2-bit field shall identify the data unit as a **TRANSFER FRAME**; it shall be set to “00”.

This Recommendation defines Version 1 of the Transfer Frame. Reference [6] defines a similar data unit which is distinguished by a different value of the Version Number.

5.1.2 TRANSFER FRAME IDENTIFICATION FIELD

- a. The **TRANSFER FRAME IDENTIFICATION FIELD** shall be contained within bits 2–15 of the **TRANSFER FRAME PRIMARY HEADER**.
- b. This 14-bit field shall be sub-divided into three sub-fields as follows:

	Length in bits
— SPACECRAFT IDENTIFIER	10
— VIRTUAL CHANNEL IDENTIFIER	3
— OPERATIONAL CONTROL FIELD FLAG	1

This field identifies the generator of the Transfer Frame, it specifies the Virtual Channel to which it belongs, and it provides information on the format of the Transfer Frame.

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5.1.2.1 SPACECRAFT IDENTIFIER

- a. Bits 2–11 of the **TRANSFER FRAME PRIMARY HEADER** shall contain the **SPACECRAFT IDENTIFIER**.
- b. The **SPACECRAFT IDENTIFIER** is assigned by CCSDS and shall provide the identification of the spacecraft which created the frame of data.
- c. The **SPACECRAFT IDENTIFIER** shall be static throughout all **MISSION PHASES**.

Different Spacecraft IDs may be assigned for normal operations and for development vehicles using the ground networks during pre-launch test operations, and for simulated data streams. The Secretariat of the CCSDS assigns Spacecraft Identifiers according to the procedures in Reference [7].

5.1.2.2 VIRTUAL CHANNEL IDENTIFIER

- a. Bits 12–14 of the **TRANSFER FRAME PRIMARY HEADER** shall contain the **VIRTUAL CHANNEL IDENTIFIER**.
- b. The **VIRTUAL CHANNEL IDENTIFIER** provides the identification of the **VIRTUAL CHANNEL**.

The order of occurrence of different Virtual Channels on a Master Channel may vary.

5.1.2.3 OPERATIONAL CONTROL FIELD FLAG

- a. Bit 15 of the **TRANSFER FRAME PRIMARY HEADER** shall contain the **OPERATIONAL CONTROL FIELD FLAG**.
- b. The **OPERATIONAL CONTROL FIELD FLAG** shall indicate the presence or absence of the **OPERATIONAL CONTROL FIELD**. It shall be “1”, if the **OPERATIONAL CONTROL FIELD** is present; it shall be “0”, if the **OPERATIONAL CONTROL FIELD** is not present.
- c. The **OPERATIONAL CONTROL FIELD FLAG** shall be static either within a specific **MASTER CHANNEL** or specific **VIRTUAL CHANNELS** throughout a **MISSION PHASE**.

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5.1.3 MASTER CHANNEL FRAME COUNT FIELD

- a. The **MASTER CHANNEL FRAME COUNT FIELD** shall be contained within bits 16–23 of the **TRANSFER FRAME PRIMARY HEADER**.
- b. This 8-bit field shall contain a sequential binary count (modulo 256) of each **TRANSFER FRAME** transmitted within a specific **MASTER CHANNEL**.
- c. A re-setting of the **MASTER CHANNEL FRAME COUNT** before reaching 255 shall not take place unless it is unavoidable.

The purpose of this field is to provide a running count of the frames which have been transmitted through the same Master Channel.

If the Master Channel Frame Count is re-set due to an unavoidable re-initialisation, the completeness of a sequence of Transfer Frames cannot be determined.

5.1.4 VIRTUAL CHANNEL FRAME COUNT FIELD

- a. The **VIRTUAL CHANNEL FRAME COUNT FIELD** shall be contained within bits 24–31 of the **TRANSFER FRAME PRIMARY HEADER**.
- b. This 8-bit field shall contain a sequential binary count (modulo 256) of each **TRANSFER FRAME** transmitted through a specific **VIRTUAL CHANNEL** of a **MASTER CHANNEL**.
- c. A re-setting of the **VIRTUAL CHANNEL FRAME COUNT** before reaching 255 shall not take place unless it is unavoidable.

The purpose of this field is to provide individual accountability for each of the maximum eight Virtual Channels, primarily to enable systematic Source Packet extraction from the Transfer Frame Data Field.

If the Virtual Channel Frame Count is re-set due to an unavoidable re-initialisation, the completeness of a sequence of Transfer Frames in the related Virtual Channel can not be determined.

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5.1.5 TRANSFER FRAME DATA FIELD STATUS FIELD

- a. The **TRANSFER FRAME DATA FIELD STATUS FIELD** shall be contained within bits 32–47 of the **TRANSFER FRAME PRIMARY HEADER**.
- b. This 16-bit field shall be sub-divided into five sub-fields as follows:

	Length in bits
— TRANSFER FRAME SECONDARY HEADER FLAG	1
— SYNCHRONISATION FLAG	1
— PACKET ORDER FLAG	1
— SEGMENT LENGTH IDENTIFIER	2
— FIRST HEADER POINTER	11

This field indicates whether a Secondary Header is present. Further, it provides information on the type of data contained in the frame and provides, together with the Virtual Channel Frame Count, the control information necessary to enable Source Packets to be extracted from the Transfer Frame Data Field.

5.1.5.1 TRANSFER FRAME SECONDARY HEADER FLAG

- a. Bit 32 of the **TRANSFER FRAME PRIMARY HEADER** shall contain the **TRANSFER FRAME SECONDARY HEADER FLAG**.
- b. The **TRANSFER FRAME SECONDARY HEADER FLAG** shall signal the presence or absence of the **TRANSFER FRAME SECONDARY HEADER**. It shall be “1”, if a **TRANSFER FRAME SECONDARY HEADER** is present; it shall be “0”, if a **TRANSFER FRAME SECONDARY HEADER** is not present.
- c. The **TRANSFER FRAME SECONDARY HEADER FLAG** shall be static within a specific **MASTER CHANNEL** throughout a **MISSION PHASE** when the **TRANSFER FRAME SECONDARY HEADER** is associated with a **MASTER CHANNEL**.
- d. The **TRANSFER FRAME SECONDARY HEADER FLAG** shall be static within a specific **VIRTUAL CHANNEL** throughout a **MISSION PHASE** when the **TRANSFER FRAME SECONDARY HEADER** is associated with a **VIRTUAL CHANNEL**.

For the significance of the above-mentioned associations see Item 5.2.d.

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5.1.5.2 SYNCHRONISATION FLAG

- a. Bit 33 of the **TRANSFER FRAME PRIMARY HEADER** shall contain the **SYNCHRONISATION FLAG**.
- b. The **SYNCHRONISATION FLAG** shall signal the type of data which are inserted into the **TRANSFER FRAME DATA FIELD**. It shall be "0", if octet-synchronised and forward-ordered **SOURCE PACKETS** or **IDLE DATA** are inserted; it shall be "1", if **PRIVATELY DEFINED DATA** are inserted.
- c. The **SYNCHRONISATION FLAG** shall be static within a specific **VIRTUAL CHANNEL** throughout a **MISSION PHASE**.

Source Packet data units are normally inserted into the Transfer Frame Data Field synchronously on octet boundaries, one following directly after another. Generally, the Source Packets "spill over" into the next frame for the same Virtual Channel; therefore, Source Packets do not usually begin at the first octet of the Transfer Frame Data Field. The location of the first Source Packet header in a particular Transfer Frame is identified by the First Header Pointer Field (see also comment under Paragraph 5.1.5.5).

If a Project chooses not to observe octet boundaries when placing Source Packet data into the Transfer Frame Data Field, the data are considered to be Privately Defined Data.

5.1.5.3 PACKET ORDER FLAG

- a. Bit 34 of the **TRANSFER FRAME PRIMARY HEADER** shall contain the **PACKET ORDER FLAG**.
- b.. If the **SYNCHRONISATION FLAG** is set to "0", the **PACKET ORDER FLAG** is reserved for future use by the CCSDS and shall be set to "0".
- c. If the **SYNCHRONISATION FLAG** is set to "1", the use of the **PACKET ORDER FLAG** is undefined.

5.1.5.4 SEGMENT LENGTH IDENTIFIER

- a. Bits 35 and 36 of the **TRANSFER FRAME PRIMARY HEADER** shall contain the **SEGMENT LENGTH IDENTIFIER**.
- b. If the **SYNCHRONISATION FLAG** is set to "0", the **SEGMENT LENGTH IDENTIFIER** shall be set to "11".

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This Identifier was required for earlier versions of this Recommendation to allow for the use of Source Packet Segments, which are no longer defined. Its value has been set to the value used to denote non-use of Source Packet Segments in previous versions.

- c. If the **SYNCHRONISATION FLAG** is set to “1”, the **SEGMENT LENGTH IDENTIFIER** is undefined.

5.1.5.5 FIRST HEADER POINTER

- a. Bits 37–47 of the **TRANSFER FRAME PRIMARY HEADER** shall contain the **FIRST HEADER POINTER**.
- b. If the **SYNCHRONISATION FLAG** is set to “0”, the **FIRST HEADER POINTER** shall contain information on the position of the first **SOURCE PACKET** within the **TRANSFER FRAME DATA FIELD**.
- c. The locations of the octets in the **TRANSFER FRAME DATA FIELD** shall be numbered in ascending order. The first octet in this field is assigned the number 0. The **FIRST HEADER POINTER** shall contain the binary representation of the location of the first octet of the first **PACKET PRIMARY HEADER**.

The locations of any subsequent headers within the same Transfer Frame Data Field will be determined by calculating these locations using the Packet Data Length Field (see Sub-Section 3.1.4).

The specification covers also the following two special cases:

- (1) *If a first Source Packet Primary Header starts at the end of the Transfer Frame Data Field within Frame N and spills over into Frame M of the same Virtual Channel, the First Header Pointer in Frame N indicates the start of this Header.*
- (2) *If a Source Packet Header is split between the Frames N and M ($M > N$), the First Header Pointer in Frame M ignores the residue of the split header and only indicates the start of any subsequent new Source Packet Header within Frame M.*

In both cases (1) and (2) above, one or more Frames with Idle Data may occur between Frame N and Frame M ($M > N$).

- d. If no **PACKET PRIMARY HEADER** starts in the **TRANSFER FRAME DATA FIELD**, the **FIRST HEADER POINTER** shall be set to “1111111111”.
- e. If a **TRANSFER FRAME** contains **IDLE DATA** in its **TRANSFER FRAME DATA FIELD**, the **FIRST HEADER POINTER** shall be set to “1111111110”.

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- f. If the **SYNCHRONISATION FLAG** is set to “1”, the **FIRST HEADER POINTER** is undefined.

5.2 **TRANSFER FRAME SECONDARY HEADER**

- a. If present, the **TRANSFER FRAME SECONDARY HEADER** shall follow, without gap, the **TRANSFER FRAME PRIMARY HEADER**.
- b. The **TRANSFER FRAME SECONDARY HEADER** is optional; its presence or absence shall be signalled by the **TRANSFER FRAME SECONDARY HEADER FLAG** in the **TRANSFER FRAME PRIMARY HEADER** (see Paragraph 5.1.5.1).
- c. The **TRANSFER FRAME SECONDARY HEADER** shall consist of an integral number of octets as follows:

	Length in bits
— TRANSFER FRAME SECONDARY HEADER IDENTIFICATION FIELD	8
— TRANSFER FRAME SECONDARY HEADER DATA FIELD	8, 16, ... or 504

- d. The **TRANSFER FRAME SECONDARY HEADER** shall be associated with either a **MASTER CHANNEL** or a **VIRTUAL CHANNEL**.

The association of a Secondary Header with a Master Channel allows data to be transferred frame-synchronously with respect to this Master Channel. See Reference [8] for a description of the services based on this field.

- e. The **TRANSFER FRAME SECONDARY HEADER** shall be of fixed length within the associated **MASTER CHANNEL** or within the associated **VIRTUAL CHANNEL** throughout a **MISSION PHASE**.

5.2.1 **TRANSFER FRAME SECONDARY HEADER IDENTIFICATION FIELD**

- a. The **TRANSFER FRAME SECONDARY HEADER IDENTIFICATION FIELD** shall be contained within bits 0–7 of the **TRANSFER FRAME SECONDARY HEADER**.
- b. The **TRANSFER FRAME SECONDARY HEADER IDENTIFICATION FIELD** shall be sub-divided into two sub-fields as follows:

	Length in bits
— the TRANSFER FRAME SECONDARY HEADER VERSION NUMBER FIELD	2
— the TRANSFER FRAME SECONDARY HEADER LENGTH FIELD	6