



AEROSPACE MATERIAL SPECIFICATION

AMS2510**REV. E**

Issued	1940-10
Revised	1980-04
Noncurrent	1996-03
Reaf. Nonc.	2013-11

Superseding AMS2510D

Engine Gray Finishing
Low Baking

RATIONALE

AMS2510E has been reaffirmed to comply with the SAE five-year review policy.

This specification has been declared "NONCURRENT" by the Aerospace Materials Division, SAE, as of March 1996. It is recommended, therefore, that this specification not be specified for new designs.

This cover sheet should be attached to the "D" revision of the subject specification.

"NONCURRENT" refers to those materials which have previously been widely used and which may be required on some existing designs in the future. The Aerospace Materials Division, however, does not recommend these as standard materials for future use in new designs. Each of these "NONCURRENT" specifications is available from SAE upon request.

PREPARED UNDER THE JURISDICTION OF AMS COMMIT-TEE "B"

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<http://www.sae.org/technical/standards/AMS2510E>**

1. SCOPE:

1.1 Purpose: This specification covers the engineering requirements for finishing aircraft parts and assemblies with an engine gray enamel.

1.2 Application: Primarily for parts and assemblies operating in service up to 350°F (175°C).

2. APPLICABLE DOCUMENTS: The following publication form a part of this specification to the extent specified herein. The latest issue of Aerospace Material Specifications (AMS) shall apply. The applicable issue of other documents shall be as specified in AMS 2350.

2.1 SAE Publications: Available from Society of Automotive Engineers, Inc., 400 Commonwealth Drive, Warrendale, PA 15096.

2.1.1 Aerospace Material Specifications:

AMS 2350 - Standard and Test Methods

AMS2400 - Cadmium Plating

AMS2470 - Anodic Treatment of Aluminum Alloys, Chromic Acid Process

AMS 2471 - Anodic Treatment of Aluminum Alloys, Sulfuric Acid Process, Undyed Coating

AMS 2475 - Protective Treatments, Magnesium Base Alloys

AMS 2480 - Phosphate Treatment, Paint Base

AMS 3110 - Primer, Zinc Chromate

AMS 3125 - Enamel, Glycexyl Phthdate, Engine Gray Baking

2.2 U.S. Government Publications: Available from Commanding Officer, Naval Publications and Forms Center, 5801 Tabor Avenue, Philadelphia, PA 19120.

2.2.1 Military Standards:

MIL-STD-794 - Parts and Equipment, Procedures for Packaging and Packing of

3. TECHNICAL REQUIREMENTS:

3.1 Preparation: Parts and assemblies, before being painted, shall be prepared as follows; elapsed time between preparatory treatments and priming shall be as short as practicable:

3.1.1 Aluminum and Aluminum Alloys: Both wrought and cast parts shall be anodized in accordance with AMS 2470 or AMS 2471.

- 3.1.1.1 Aluminum Assemblies: Aluminum alloy parts which with parts made of other materials constitute assemblies shall be anodized in accordance with AMS 2470 or AMS 2471 before assembling with such other parts. Anodizing in accordance with AMS 2470 should be used on assemblies in which sulfuric acid could be trapped. Assemblies with parts made of other metals may be anodized if such other parts are insulated.
- 3.1.2 Magnesium Alloys: Both wrought and cast parts shall be treated in accordance with AMS 2475. Machining of external surfaces shall be completed prior to such treatment.
- 3.1.3 Steel: Cadmium plated parts shall be thoroughly netralized in accordance with AMS 2400. Unplated parts shall be cleaned free from oil, grease, dirt, and rust and then treated in accordance with AMS 2480.
- 3.2 Procedure:
- 3.2.1 Priming: One coat of AMS 3110 zinc chromate primer shall be applied to all surfaces of metallic parts requiring enameling, except as follows:
- 3.2.1.1 Anodized rivets shall not be primed as details.
- 3.2.1.2 Two coats of primer shall be applied to magnesium alloy parts.
- 3.2.2 Cleaning: When there are intervening operations between priming and enameling, such as assembling or additional machining, the parts shall be thoroughly cleaned by spraying with clean naphtha, or other solvent of low volatility, and given another coat of AMS 3110 zinc chromate primer before the first coat of enamel is applied.
- 3.2.3 Baking: Each coat of primer shall be baked at 250° - 300° F (120° - 150°C) unless baking at a lower temperature or air drying is permitted by purchaser.
- 3.2.4 Enameling: The following requirements apply to all metallic surfaces which are exposed after parts are assembled in the power plant or aircraft, unless otherwise specified. Magnesium alloy parts used as covers but not in contact with oil shall be enameled on both inside and outside surfaces except on contacting machined surfaces. Enameling of spot faces is optional.
- 3.2.4.1 Three coats of AMS 3125 gray enamel shall be applied to magnesium alloy parts for use on other than aircraft power plants.
- 3.2.4.2 Two coats of AMS 3125 gray enamel shall be applied to magnesium alloy parts for use on aircraft power plants and to parts of all other metals for all applications.
- 3.2.4.3 Each coat of enamel shall be thoroughly baked at not higher than 300° F (150° C) or preliminary coats may be air-dried dust-free and the final coat baked firm and hard at not higher than 300° F (150°C).
- 3.2.4.4 Enamel is not required on nonmetallic materials and associated metal parts, on threaded sections, or on removable steel parts, unless otherwise specified.
- 3.3 Quality: Finish shall be smooth uniform in color and gloss, and free from pinholes, sags, runs, heavy edges, foreign materials, and other imperfections detrimental to appearance or performance of parts.