



AEROSPACE MATERIAL SPECIFICATION

Society of Automotive Engineers, Inc.
400 COMMONWEALTH DRIVE, WARRENDALE, PA. 15096

AMS 4310

Issued 7-1-76
Revised

UNS A97075

ALUMINUM ALLOY RINGS, ROLLED OR FORGED
5.6Zn - 2.5Mg - 1.6Cu - 0.26Cr (7075-T651, 7075-T652)
Mechanically Stress Relieved

1. SCOPE:

1.1 Form: This specification covers an aluminum alloy in the form of rolled or forged rings.

1.2 Application: Primarily for structural machined parts where good stability is required during machining. Not recommended for fusion welding. Certain design and fabricating procedures may cause this material to be susceptible to stress-corrosion cracking; ARP 823 recommends practices to minimize such conditions.

1.3 Classification: The rings covered by this specification are classified by type of mechanical stress relief as follows:

Type 1 - Stress-relieved by stretching (7075-T651)

Type 2 - Stress-relieved by compression (7075-T652)

1.3.1 Either type may be supplied, unless a specific type is ordered.

2. APPLICABLE DOCUMENTS: The following publications form a part of this specification to the extent specified herein. The latest issue of Aerospace Material Specifications (AMS) and Aerospace Recommended Practices (ARP) shall apply. The applicable issue of other documents shall be as specified in AMS 2350.

2.1 SAE Publications: Available from Society of Automotive Engineers, Inc., 400 Commonwealth Drive, Warrendale, PA 15096.

2.1.1 Aerospace Material Specifications:

AMS 2350 - Standards and Test Methods

AMS 2375 - Approval and Control of Critical Forgings

AMS 2808 - Identification, Forgings

2.1.2 Aerospace Recommended Practices:

ARP 823 - Minimizing Stress Corrosion in Wrought Heat Treatable Aluminum Alloy Products

2.2 ASTM Publications: Available from American Society for Testing and Materials, 1916 Race Street, Philadelphia, PA 19103.

ASTM B557 - Tension Testing Wrought and Cast Aluminum and Magnesium Alloy Products

ASTM E10 - Brinell Hardness of Metallic Materials

ASTM E34 - Chemical Analysis of Aluminum and Aluminum-Base Alloys

2.3 Government Publications: Available from Commanding Officer, Naval Publications and Forms Center, 5801 Tabor Avenue, Philadelphia, PA 19120.

2.3.1 Federal Standards:

Federal Test Method Standard No. 151 - Metals; Test Methods

SAE Technical Board rules provide that: "All technical reports, including standards approved by the Board, are advisory only. Their use by anyone engaged in industry or trade is entirely voluntary. There is no agreement to adhere to any SAE standard or recommended practice, and no commitment to conform to or be guided by any technical report. In formulating and approving technical reports, the Board and its committees will not investigate or consider patents which may apply to the subject matter. Prospective users of the report are responsible for protecting themselves against infringement of patents."

2.3.2 Military Specifications:

MIL-H-6088 - Heat Treatment of Aluminum Alloys
 MIL-I-8950 - Inspection, Ultrasonic, Wrought Metals, Process for

2.3.3 Military Standards:

MIL-STD-649 - Aluminum and Magnesium Products, Preparation for Shipment and Storage

3. TECHNICAL REQUIREMENTS:

- 3.1 Composition: Shall conform to the following percentages by weight, determined by wet chemical methods in accordance with ASTM E34, by spectrographic methods in accordance with Federal Test Method Standard No. 151, Method 112, or by other approved analytical methods:

	min	max
Zinc	5.1	6.1
Magnesium	2.1	2.9
Copper	1.2	2.0
Chromium	0.18	0.35
Iron	--	0.50
Silicon	--	0.40
Manganese	--	0.30
Titanium + Zirconium	--	0.25
Titanium	--	0.20
Other Impurities, each	--	0.05
Other Impurities, total	--	0.15
Aluminum	remainder	

- 3.2 Condition: Rings shall be supplied in the following condition; heat treatments shall be performed in accordance with MIL-H-6088:

- 3.2.1 Type 1: Solution heat treated, stress-relieved by stretching to produce a permanent set of 1 - 5%, and precipitation heat treated.
- 3.2.2 Type 2: Solution heat treated, stress-relieved by compression to produce a permanent set of 1 - 5%, and precipitation heat treated. During compression, primary forces shall be applied in the axial direction and on individual rings approximately final dimensions.

- 3.3 Properties: Rings shall conform to the following requirements:

3.3.1 Tensile Properties:

- 3.3.1.1 Rings with OD to Wall Thickness Ratio Less than 10: Shall be as agreed upon by purchaser and vendor.

- 3.3.1.2 Rings with OD to Wall Thickness Ratio of 10 or Greater: Shall be in accordance with Table I, determined in accordance with ASTM B557. Tensile tests are not required in any direction from which a specimen at least 2.375 in. (60.32 mm) in length cannot be obtained.

TABLE I

Nominal Thickness at Time of Heat Treatment Inches (See 3.3.1.2.1)	Specimen Orientation (See 3.3.1.2.2)	Tensile Strength psi, min	Yield Strength at 0.2% Offset psi, min	Elongation in 2 in. or 4D %, min
Up to 2, incl	Tangential	74,000	63,000	9
	Axial	73,000	61,000	4
Over 2 to 3, incl	Tangential	73,000	61,000	9
	Axial	71,000	59,000	4
	Radial	69,000	57,000	2
Over 3 to 4, incl	Tangential	71,000	60,000	8
	Axial	70,000	58,000	3
	Radial	68,000	56,000	1
Over 4 to 5, incl	Tangential	69,000	58,000	7
	Axial	68,000	56,000	3
	Radial	66,000	55,000	1
Over 5 to 6, incl	Tangential	68,000	56,000	6
	Axial	66,000	55,000	3
	Radial	65,000	54,000	1

TABLE I (SI)

Nominal Thickness at Time of Heat Treatment Millimetres (See 3.3.1.2.1)	Specimen Orientation (See 3.3.1.2.2)	Tensile Strength MPa, min	Yield Strength at 0.2% Offset MPa, min	Elongation in 50.8 mm or 4D %, min
Up to 51, incl	Tangential	510	434	9
	Axial	503	421	4
Over 51 to 76, incl	Tangential	503	421	9
	Axial	490	407	4
	Radial	476	393	2
Over 76 to 102, incl	Tangential	490	414	8
	Axial	483	400	3
	Radial	469	386	1
Over 102 to 127, incl	Tangential	476	400	7
	Axial	469	386	3
	Radial	455	379	1
Over 127 to 152, incl	Tangential	469	386	6
	Axial	455	379	3
	Radial	448	372	1

- 3.3.1.2.1 Thickness is defined as the smaller of the wall thickness (one-half the difference between nominal OD and nominal ID) and height (axial) dimensions.
- 3.3.1.2.2 Tangential test requirements apply to specimens machined with axis of specimen tangential to the ring OD (Parallel to the direction of rolling). Axial test requirements apply to specimens machined with axis of specimen parallel to the axis of the ring (long transverse to the direction of rolling). Radial test requirements apply to specimens machined with axis of specimen parallel to the radius of the ring (short transverse to the direction of rolling). All specimens shall be machined from the core of the ring.

- 3.3.2 Hardness: Should be not lower than 135 HB/10/500, 135 HB/14.3/1000, or 140 HB/10/1000, determined in accordance with ASTM E10, but the rings shall not be rejected on the basis of hardness if the tensile property requirements are met.
- 3.4 Quality: Rings, as received by the purchaser, shall be uniform in quality and condition, sound, and free from foreign materials and from internal and external imperfections detrimental to usage of the rings.
- 3.4.1 Each ring shall be ultrasonically inspected in accordance with MIL-I-8950, unless otherwise specified, and shall meet the Class A acceptance limits of that specification.
4. QUALITY ASSURANCE PROVISIONS:
- 4.1 Responsibility for Inspection: The vendor of rings shall supply all samples and shall be responsible for performing all required tests. Results of such tests shall be reported to the purchaser as required by 4.5. Purchaser reserves the right to perform such confirmatory testing as he deems necessary to ensure that the rings conform to the requirements of this specification.
- 4.2 Classification of Tests: Tests to determine conformance to all technical requirements of this specification are classified as acceptance tests.
- 4.2.1 For direct U.S. Military procurement, substantiating test data and, when requested, preproduction test material shall be submitted to the cognizant agency as directed by the procuring activity, the contracting officer, or the request for procurement.
- 4.3 Sampling: Shall be as follows; a lot shall be all rings of the same size solution heat treated in the same batch furnace load or consecutively in a continuous furnace in an 8-hr period and precipitation heat treated as a unit:
- 4.3.1 Composition: At least one sample shall be taken by the producer from each group of ingots poured simultaneously from the same source of molten metal.
- 4.3.1.1 Unless compliance with 4.3.1 is established, an analysis shall be made for each 4000 lb (1816kg) or less of material comprising the lot except that not more than one analysis shall be required per piece.
- 4.3.2 Tensile Properties: Except when testing in one or more directions is not required by 3.3.1, test specimens in the tangential, axial, and radial directions shall be taken from a ring, ring prolongation, or ring segment representing the lot. When ring segments are used for testing, the segments shall be cut from a ring which has been solution heat treated and stress-relieved with the production rings. Solution heat treated and stress-relieved ring segments shall be included in each precipitation heat treatment furnace load.
- 4.3.2.1 When requested by purchaser, at least one half of each ring segment obtained as in 4.3.2 or one half of each ring prolongation tested shall be submitted to the purchaser with the rings represented.
- 4.3.3 Hardness: Each ring. If hardness of any ring indicates low tensile properties, the ring having the lowest hardness shall be tested for tensile properties.
- 4.4 Approval: When specified, approval and control of rings shall be in accordance with AMS 2375.
- 4.5 Reports:
- 4.5.1 The vendor of rings shall furnish with each shipment three copies of a report stating that the chemical composition conforms to the requirements of this specification and showing the results of tests on each lot to determine conformance to the other technical requirements of this specification. This report shall include the purchase order number, material specification number, size or part number, and quantity.