



AEROSPACE MATERIAL SPECIFICATION

Society of Automotive Engineers, Inc.
TWO PENNSYLVANIA PLAZA, NEW YORK, N. Y. 1000

AMS 6356A

Superseding AMS 6356

Issued 1-15-62

Revised 11-1-68

STEEL SHEET, STRIP, AND PLATE 0.95Cr - 0.20Mo (0.30 - 0.35C)

1. **ACKNOWLEDGMENT:** A vendor shall mention this specification number and its revision letter in all quotations and when acknowledging purchase orders.
2. **APPLICATION:** Primarily for heat treated parts which require through hardening to minimum tensile strength of 180,000 psi in sections up to 0.375 in. thick and proportionately lower strength in heavier thicknesses and which may require welding during fabrication.
3. **COMPOSITION:**

	min	max
Carbon	0.30	0.35
Manganese	0.40	0.60
Silicon	0.20	0.50
Phosphorus	--	0.025
Sulfur	--	0.025
Chromium	0.80	1.10
Molybdenum	0.15	0.25
Nickel	--	0.25
Copper	--	0.35

- 3.1 **Check Analysis:** Composition variations shall meet the requirements of the latest issue of AMS 2259, paragraph titled "Low Alloy Steels."
4. **CONDITION:** Unless otherwise ordered, the product shall be supplied in the following condition:
 - 4.1 **Sheet and Strip:** Cold finished and bright annealed; or hot rolled, annealed if necessary, and descaled; having hardness not higher than Rockwell B 95 or equivalent.
 - 4.2 **Plate:** Hot rolled, annealed if necessary, and descaled, having hardness not higher than Rockwell B 95 or equivalent.
 - 4.3 If material is ordered spheroidized, the degree of spheroidization shall be as agreed upon by purchaser and vendor.
5. **TECHNICAL REQUIREMENTS:** When ASTM methods are specified for determining conformance to the following requirements, tests shall be conducted in accordance with the issue of the ASTM method listed in the latest issue of AMS 2350.
 - 5.1 **Grain Size:** Predominantly 5 or finer with occasional grains as large as 3 permissible, ASTM E112, McQuaid-Ehn test.
 - 5.2 **Decarburization:**
 - 5.2.1 **Material Under 0.045 In. in Thickness:** The method of test and the allowance shall be as agreed upon by purchaser and vendor.
 - 5.2.2 **Material 0.045 In. and Over in Thickness:**

- 5.2.2.1 Specimens: Shall be the full thickness of the material except that specimens from plate over 0.250 in. thick shall be slices approximately 0.250 in. thick cut parallel to and preserving one original surface of the plate. Recommended specimen size is 1 x 4 inches.
- 5.2.2.2 Procedure: Specimens shall be hardened by austenitizing and quenching; preferably they shall not be tempered but, if tempered, the tempering temperature shall be not higher than 300 F (149 C). During heat treatment, specimens shall be protected by suitable atmosphere or medium or by suitable plating to prevent carburization or further decarburization. Protective plating, if used, shall then be removed from specimens of material 0.045 to 0.250 in., excl, in thickness and a portion of the specimen shall be step ground to a depth of 0.050 in. or half thickness, whichever is less. Specimens from material 0.250 in. and over in thickness shall be ground to remove from the original surface of the plate the amount of metal shown below and a portion of the specimen shall be further ground to a depth of at least 1/3 the original thickness of the specimen. At least three Rockwell hardness readings shall be taken on each prepared step and each group of readings averaged.

Nominal Original Thickness Inches	Surface Depth Removal Inch
0.250 to 0.375, incl	0.020
Over 0.375 to 0.500, incl	0.025
Over 0.500 to 0.750, incl	0.030
Over 0.750 to 1.000, incl	0.035
Over 1.000 to 2.000, incl	0.040

5.2.2.3 Allowance:

- 5.2.2.3.1 Material 0.045 to 0.250 In., Excl, Thick: Unless otherwise specified, the product shall be free from complete decarburization. It shall also be free from partial decarburization to the extent that the difference in hardness between the surface and the nondecarburized depth below the surface shall be not greater than 2 points on the Rockwell A scale.

- 5.2.2.3.2 Material 0.250 In. and Over Thick: The difference in hardness between the two prepared steps shall be not greater than 2 points on the Rockwell C scale.

- 5.3 Hardenability: The steel shall have hardenability of J57=1 max and J46=3 min when determined in accordance with the Method of Determining Hardenability published in the latest issue of the SAE Handbook, except that the steel shall be normalized at $1700\text{ F} \pm 10$ ($926.7\text{ C} \pm 5.6$) and the test specimen austenitized at $1600\text{ F} \pm 10$ ($871.1\text{ C} \pm 5.6$). The hardenability test is not required on a product which will not yield a suitable specimen, but the steel from which the product is made shall conform to the hardenability specified in this paragraph.

- 5.4 Bending: Material shall withstand, without cracking, bending at room temperature through the angle indicated below around a diameter equal to the nominal thickness of the material, with axis of bend parallel to the direction of rolling.

Nominal Thickness Inch	Angle deg, min
Up to 0.249, incl	180
Over 0.249 to 0.749, incl	90

6. QUALITY: Steel shall be aircraft quality and shall conform to the latest issue of AMS 2301. The product shall be uniform in quality and condition, clean, sound, and free from foreign materials and from internal and external imperfections detrimental to fabrication or to performance of parts.
7. TOLERANCES: Unless otherwise specified, tolerances shall conform to all applicable requirements of the latest issue of AMS 2252; for strip, tolerances for cold finished shall apply.