



400 Commonwealth Drive, Warrendale, PA 15096-0001

AEROSPACE STANDARD

SAE AS7468

REV.
A

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Submitted for recognition as an American National Standard

Superseding AS7468

BOLTS, COBALT-CHROMIUM-NICKEL ALLOY UNS R30035, Tensile Strength 260 ksi, Procurement Specification

FSC 5306

1. SCOPE:

1.1 Type:

This procurement specification covers bolts made from a corrosion resistant, work-strengthened, cobalt-chromium-nickel alloy of the type identified under the Unified Numbering System as UNS R30035. The following specification designations and their properties are covered:

AS7468	260 ksi minimum ultimate tensile strength at room temperature 135 ksi tension to 13.5 ksi tension fatigue at room temperature
AS7468-1	260 ksi minimum ultimate tensile strength at room temperature 145 ksi minimum ultimate shear strength at room temperature

1.2 Application:

Primarily for aerospace propulsion system applications where an excellent combination of strength, fatigue, and corrosion resistance is required at temperatures up to 700 °F.

1.3 Safety - Hazardous Materials:

While the materials, methods, applications, and processes described or referenced in this specification may involve the use of hazardous materials, this specification does not address the hazards which may be involved in such use. It is the sole responsibility of the user to ensure familiarity with the safe and proper use of any hazardous materials and to take necessary precautionary measures to ensure the health and safety of all personnel involved.

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SAE AS7468 Revision A**2. REFERENCES:****2.1 Applicable Documents:**

The following publications form a part of this document to the extent specified herein. The latest issue of SAE publications shall apply. The applicable issue of other publications shall be the issue in effect on the date of the purchase order. In the event of conflict between the text of this document and references cited herein, the text of this document takes precedence. Nothing in this document, however, supersedes applicable laws and regulations unless a specific exemption has been obtained.

2.1.1 SAE Publications: Available from SAE, 400 Commonwealth Drive, Warrendale, PA 15096-0001.

- AMS 2750 Pyrometry
 AMS 5844 Alloy, Corrosion Resistant, Bars 20Cr 35Ni 35Co 10Mo, Vacuum Induction Plus Vacuum Consumable Electrode Melted, Solution Heat Treated and Work Strengthened
- AS3062 Bolts, Screws, and Studs, Screw Thread Requirements
 AS3063 Bolts, Screws, and Studs, Geometric Control Requirements

2.1.2 U.S. Government Publications: Available from DODSSP, Subscription Services Desk, Building 4D, 700 Robbins Avenue, Philadelphia, PA 19111-5094.

- MIL-S-8879 Screw Threads, Controlled Radius Root With Increased Minor Diameter; General Specification For
- MIL-STD-1312-6 Fastener Test Methods, Method 6, Hardness
 MIL-STD-1312-8 Fastener Test Methods, Method 8, Tensile Strength
 MIL-STD-1312-11 Fastener Test Methods, Method 11, Tension Fatigue
 MIL-STD-1312-13 Fastener Test Methods, Method 13, Double Shear Test
- MS21296 Bolt, Tension, Steel, 260 ksi Ft_u, 450 °F, External Wrenching Spline Drive, Flanged Head

2.1.3 ASTM Publications: Available from ASTM, 100 Barr Harbor Drive, West Conshohocken, PA 19428-2959.

- ASTM E 8 Tension Testing of Metallic Materials
 ASTM E 112 Determining Average Grain Size
 ASTM E 140 Standard Hardness Tables for Metals
 ASTM E 1417 Liquid Penetrant Examination
 ASTM D 3951 Commercial Packaging

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2.1.4 ASME Publication: Available from ASME, 22 Law Drive, Box 2900, Fairfield, NJ 07007-2900.

ASME B46.1 Surface Texture (Surface Roughness, Waviness, and Lay)

2.2 Definitions:

BURR: A rough edge or ridge left on the metal due to a cutting, grinding, piercing, or blanking operation.

COLD ROLLING: Forming material below the recrystallation temperature.

CRACK: Rupture in the material which may extend in any direction and which may be intercrystalline or transcrystalline in character.

DEFECT: Any nonconformance of the unit of product with specified requirements.

DEFECTIVE: A unit of product which contains one or more defects.

DISCONTINUITY: An interruption in the normal physical structure or configuration of a part; such as a lap, seam, inclusion, crack, machining tear, or stringer.

HEAT PATTERN: A discernible difference in etched appearance between the head and shank caused by the plastic forming of the head.

INCLUSION: Nonmetallic particles originating from the material making process. They may exist as discrete particles or strings of particles extending longitudinally.

LAP: Surface imperfection caused by folding over metal fins or sharp corners and then rolling or forging them into the surface. The allowable lap depth shall not exceed the limit specified herein. The minimum condition that shall be rated as a lap is a fold having its length equal to or greater than three times its width with a depth of 0.0005 inch when viewed at 200X magnification.

MACHINING TEAR: A pattern of short, jagged individual cracks, generally at right angles to the direction of machining, frequently the result of improperly set cutting tools, or dull cutting tools.

PRODUCTION INSPECTION LOT: Shall be all finished parts of the same part number, made from a single heat of alloy, heat treated at the same time to the same specified condition, produced as one continuous run, and submitted for vendor's inspection at the same time.

SEAM: Longitudinal surface imperfection in the form of an unwelded, open fold in the material.

STRINGER: A solid nonmetallic impurity in the metal bar, often the result of inclusions that have been extended during the rolling process.

TIGHT BURR: A burr closely compacted and binding in the periphery of the part without any loose ends and is within the dimensional limits of the part.

SAE AS7468 Revision A**2.3 Unit Symbols and Abbreviations:**

- ° - degree, angle
- °F - degree Fahrenheit
- HRC - hardness Rockwell C scale
- % - percent (1% = 1/100)
- lbf - pound force
- ksi - kips (1000 pounds) per square inch
- sp gr - specific gravity

3. TECHNICAL REQUIREMENTS:**3.1 Material:**

Shall be AMS 5844 alloy heading stock.

3.2 Design:

Finished (completely manufactured) parts shall conform to the following requirements:

- 3.2.1 Dimensions: The dimensions of finished parts, after all processing, including plating, shall conform to the part drawing. Dimensions shall apply after plating but before coating with dry film lubricants.
- 3.2.2 Surface Texture: Surface texture of finished parts, prior to plating or coating, shall conform to the requirements as specified on the part drawing, determined in accordance with ASME B46.1.
- 3.2.3 Threads: Screw thread UNJ profile and dimensions shall be in accordance with MIL-S-8879, unless otherwise specified on the part drawing.
 - 3.2.3.1 Incomplete Lead and Runout Threads: Incomplete threads are permissible at the entering end and the juncture of the unthreaded portion of the shank or adjacent to the head as specified in AS3062.
 - 3.2.3.2 Chamfer: The entering end of the thread shall be chamfered as specified on the part drawing.
- 3.2.4 Geometric Tolerances: Part features shall be within the geometric tolerances specified on the part drawing and, where applicable, controlled in accordance with AS3063.

SAE AS7468 Revision A**3.3 Fabrication:**

- 3.3.1 Blanks:** Heads shall be formed by hot or cold forging; machined heads are not permitted, except lightening holes may be produced by any suitable method. Wrenching recesses may be forged or machined. Flash or chip clearance in machined recesses shall not cause recess dimensions to exceed the specified limits.
- 3.3.2 Heat Treatment:** Before finishing the shank and the bearing surface of the head, cold rolling the head-to-shank fillet radius, and rolling the threads, headed blanks shall be aged as follows:
- 3.3.2.1 Heating Equipment:** Furnaces may be any type ensuring uniform temperature throughout the parts being heated and shall be equipped with, and operated by, automatic temperature controllers and data recorders conforming to AMS 2750.
- 3.3.2.2 Aging Heat Treatment:** Headed blanks shall be aged by heating to a temperature within the range 1000 to 1200 °F, holding at the selected temperature within ± 25 °F for 4 to 4.5 hours, and cooling at a rate equivalent to air cooling.
- 3.3.3 Oxide Removal:** Surface oxide and oxide penetration resulting from prior heat treatment shall be removed from the full body diameter and bearing surface of the head of the solution heat treated, work-strengthened, and aged blanks prior to cold rolling the under head fillet radius and rolling the threads. The oxide removal process shall produce no intergranular attack or corrosion of the blanks. The metal removed from the bearing surface of the head and the full body diameter of the shank shall be as little as practicable to obtain a clean, smooth surface.
- 3.3.4 Cold Rolling of Fillet Radius:** After removal of oxide as in 3.3.3, the head-to-shank fillet radius of headed parts having the radius complete throughout the circumference of the part shall be cold rolled sufficiently to remove all visual evidence of grinding or tool marks. Distortion due to cold rolling shall conform to Figure 2, unless otherwise specified on the part drawing. It shall not raise metal more than 0.002 inch above the contour at "A" or depress metal more than 0.002 inch below the contour at "B" as shown in Figure 2; distorted areas shall not extend beyond "C" as shown in Figure 2. In configurations having an undercut connected with the fillet radius, the cold rolling will be required only for 90° of fillet arc, starting at the point of tangency of the fillet radius and the bearing surface of the head. For shouldered bolts, having an unthreaded shank diameter larger than the thread major diameter and having an undercut connected with a fillet between the threaded shank and the shoulder of the unthreaded shank, the cold rolling will be required only for 90° of fillet arc, starting at the point of tangency of the fillet radius and the shouldered surface of the unthreaded shank. The shank diameter on close tolerance full shank bolts shall not exceed its maximum diameter limit after cold rolling the head-to-shank fillet radius.
- 3.3.5 Thread Rolling:** Threads shall be formed on the finished blanks by a single cold rolling process after removal of oxide as in 3.3.3.

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3.3.6 Cleaning: Parts, after finishing, shall be degreased and submerged in one of the following solutions for the time and temperature shown:

- a. One volume of nitric acid (sp gr 1.42) and 9 volumes of water for not less than 20 minutes at room temperature.
- b. One volume of nitric acid (sp gr 1.42) and 4 volumes of water for 30 to 40 minutes at room temperature.
- c. One volume of nitric acid (sp gr 1.42) and 4 volumes of water for 10 to 15 minutes at 140 to 160 °F.

3.3.6.1 Water Rinse: Immediately after removal from the cleaning solution, the parts shall be thoroughly rinsed in (70 to 200 °F) water.

3.4 Product Marking:

Each part shall be identification marked as specified by the part drawing. The markings may be formed by forging or stamping, raised or depressed 0.010 inch maximum, with rounded root form on depressed characters.

3.5 Plating or Coating:

Where required, surfaces shall be plated or coated as specified by the part drawing. Plating thickness determined in accordance with plating specification.

3.6 Mechanical Properties:

Where AS7468 is specified, parts shall conform to the requirements of 3.6.1, 3.6.2, and 3.6.3. Where AS7468-1 is specified, parts shall conform to the requirements of 3.6.1, 3.6.2, and 3.6.4. Threaded members of gripping fixtures for tensile, and fatigue tests shall be of sufficient size and strength to develop the full strength of the part without stripping the thread.

AS7468 finished parts shall be tested in accordance with the following test methods:

- | | |
|---|-----------------|
| a. Hardness: | MIL-STD-1312-6 |
| b. Ultimate Tensile Strength at Room Temperature: | MIL-STD-1312-8 |
| c. Fatigue Strength at Room Temperature: | MIL-STD-1312-11 |

AS7468-1 finished parts shall be tested in accordance with the following test methods:

- | | |
|---|-----------------|
| a. Hardness: | MIL-STD-1312-6 |
| b. Ultimate Tensile Strength at Room Temperature: | MIL-STD-1312-8 |
| c. Ultimate Shear Strength at Room Temperature: | MIL-STD-1312-13 |

SAE AS7468 Revision A**3.6.1 Ultimate Tensile Strength at Room Temperature:**

3.6.1.1 Finished Parts: Tension bolts, such as double hexagon and spline drive heads, shall have an ultimate tensile load not lower than that specified in Table 2 and shall be tested to failure in order to observe fracture location, first measuring and recording the maximum tensile load achieved. If the size or shape of the bolt is such that failure would occur outside the threaded section but the bolt can be tested satisfactorily, such as bolts having a shank diameter equal to or less than the thread root diameter or having an undercut, parts shall have an ultimate tensile strength not lower than 260 ksi; for such parts, the diameter of the area on which stress is based shall be the actual measured minimum diameter of the part. Tension fasteners with either standard double hexagon or spline drive heads having a minimum metal condition in the head equal to the design parameters specified in MS21296 shall not fracture in the head-to-shank fillet radius except when this radius is connected with an undercut or with a shank diameter less than the minimum pitch diameter of the thread.

3.6.1.2 Machined Test Specimens: If the size or shape of the part is such that a tensile test cannot be made on the part, tensile tests shall be conducted in accordance with ASTM E 8 on specimens as in 4.5.7. Specimens may be required by the purchaser to perform confirmatory tests. Such specimens shall meet the following room temperature requirements:

- a. Ultimate Tensile Strength, minimum: 260 ksi
- b. Yield Strength at 0.2% Offset, minimum: 230 ksi
- c. Elongation in 4D, minimum: 8%
- d. Reduction of Area, minimum: 35%

3.6.1.2.1 When permitted by purchaser, hardness tests on the end of parts may be substituted for tensile tests of machined specimens.

3.6.2 Hardness: Shall be uniform and not lower than 44 HRC, but hardness of the threaded section and of the head-to-shank fillet area may be higher than other areas as a result of the cold rolling operations. Parts shall not be rejected on the basis of hardness if the tensile strength properties specified in 3.6.1 are met.

3.6.3 Fatigue Strength: Finished tension bolts tested in tension-tension fatigue at room temperature with maximum load as specified in Table 2 and minimum load equal to 10% of maximum load shall have average life of not less than 65,000 cycles with no part having life less than 45,000 cycles. Tests need not be run beyond 130,000 cycles. Life of parts which do not fail in less than 130,000 cycles shall be taken as 130,000 cycles for purposes of computing average life. If the shank diameter of the bolt is less than the minimum pitch diameter of thread, bolts shall withstand fatigue testing as above using loads sufficient to produce a maximum stress of 135 ksi and a minimum stress of 13.5 ksi. The above requirements apply to tension bolts, such as double hex and spline drive heads per design parameters specified in MS21296, 0.190 inch and larger in nominal thread size, having a head-to-shank fillet radius equal to or larger than that specified in MS21296, and not having an undercut; for all parts to which the above requirements do not apply, fatigue test requirements shall be as specified on the part drawing.

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3.6.4 Ultimate Shear Strength: Finished bolts having a close toleranced full shank as in MS21296 shall have an ultimate double shear load not lower than that specified in Table 2. The double shear test may be discontinued without a complete shear failure after the ultimate shear load has been reached, first measuring and recording the maximum double shear load achieved. Shear bolts having special shank diameters shall have the minimum ultimate double shear load based on 145 ksi minimum shear strength. Shear test are not required for the following conditions:

- a. Bolts threaded to head.
- b. Protruding head bolts having coarse toleranced full shank.
- c. Protruding head bolts having PD or relieved shank.

3.7 Quality:

Parts shall be uniform in quality and condition, free from burrs (tight burrs may be acceptable if part performance is not affected), foreign materials, and from imperfections detrimental to the usage of the part.

3.7.1 Macroscopic Examination, Headed Blank: A specimen cut from a headed blank shall be etched in a suitable etchant and examined at a magnification of 20X to determine conformance to the requirements of 3.7.1.1 and 3.7.1.2. The head and shank section shall extend not less than $D/2$ from the bearing surface of the head, where "D" is the nominal diameter of the shank after heading.

3.7.1.1 Flow Lines: After heading and prior to heat treatment, examination of an etched section taken longitudinally through the blank shall show flow lines or heat pattern in the shank, head-to-shank fillet, and bearing surface which follow the contour of the blank as shown in Figure 1 or Figure 1A. Flow lines or heat pattern in headed blanks having special heads, such as Dee- or Tee-shaped heads, or thinner than MS21296 standard heads, shall be as agreed upon by purchaser and vendor.

3.7.1.2 Internal Defects: Examination of longitudinal sections of the head and shank shall reveal no cracks, laps, or porosity.

3.7.2 Microscopic Examination, Finished Parts: Specimens cut from parts shall be polished, etched in Kalling's reagent, Marble's reagent, or other suitable etchant, and examined at 100X magnification to determine conformance to the requirements of 3.7.2.1, 3.7.2.2, 3.7.2.3, 3.7.2.4, 3.7.2.5, and 3.7.2.6.

3.7.2.1 Flow Lines: Examination of a longitudinal section through the threaded portion of the shank shall show evidence that the threads were rolled. See Figure 3.

3.7.2.2 Internal Defects: Examination of longitudinal sections of the head and shank shall reveal no cracks, laps, or porosity. Thread imperfections shall conform to the requirements of 3.7.2.6.

3.7.2.3 Microstructure: Parts shall have distorted grain structure indicative of cold worked material free from recrystallization in areas other than the head.

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- 3.7.2.4 Grain Size: The grain size in areas other than the head, shall be ASTM No. 4 or finer, with up to 15% of the area as large as ASTM No. 2 as determined in accordance with ASTM E 112. In case of disagreement on average grain size by the comparison method, the intercept (Heyn) procedure shall be used.
- 3.7.2.5 Surface Hardening: Parts shall have no change in hardness from core to surface except as produced during cold rolling of the head-to-shank fillet radius and during rolling of threads. In case of dispute over results of the microscopic examination, microhardness testing shall be used as a referee method; a Vickers hardness reading of an unrolled surface which exceeds the reading in the core by more than 30 points shall be evidence of nonconformance to this requirement.
- 3.7.2.6 Threads:
- 3.7.2.6.1 Root defects such as laps, seams, notches, slivers, folds, roughness, and oxide scale are not permissible (see Figure 4).
- 3.7.2.6.2 Multiple laps on the flanks of threads are not permissible regardless of location.
- 3.7.2.6.3 Single Lap on Thread Profile: Shall conform to the following:
- Thread Flank Above the Pitch Diameter: A single lap is permissible along the flank of the thread above the pitch line on either the pressure or nonpressure flank (one lap at any cross-section through the thread) provided it extends toward the crest and generally parallel to the flank (see Figure 5). The lap depth shall not exceed the limit specified in Table 1 for the applicable thread pitch. A lap extending toward the root is not permissible (see Figure 6).
 - Thread Flank Below the Pitch Diameter: A lap along the thread flank below the pitch line, regardless of direction it extends, is not permissible (see Figure 7).
 - Crest craters, crest laps, or a crest lap in combination with a crest crater are permissible provided that the imperfections do not extend deeper than the limit specified in Table 1 as measured from the thread crest when the thread major diameter is at minimum size (see Figure 8). The major diameter of the thread shall be measured prior to sectioning. As the major diameter of the thread approaches maximum size, values for depth of crest crater and crest lap imperfections listed in Table 1 may be increased by one-half of the difference between the minimum major diameter and actual major diameter as measured on the part.

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3.7.3 Fluorescent Penetrant Inspection: Prior to any required plating or coating, parts shall be subject to fluorescent penetrant inspection in accordance with ASTM E 1417, Type I, Sensitivity Level 2 minimum.

3.7.3.1 The following conditions shall be cause for rejection of parts inspected:

3.7.3.1.1 Discontinuities transverse to grain flow (i.e., at an angle of more than 10° to the axis of the shank), such as grinding checks and cracks.

3.7.3.1.2 Longitudinal indications (i.e., at an angle of 10° or less to the axis of the shank) due to imperfections other than seams, forming laps, and nonmetallic inclusions.

3.7.3.2 The following conditions shall be considered acceptable on parts inspected:

3.7.3.2.1 Parts having longitudinal indications (i.e., at an angle of 10° or less to the axis of the shank) of seams and forming laps parallel to the grain flow that are within the limits specified in 3.7.3.2.2 through 3.7.3.2.5 provided the separation between indications is not less than 0.062 inch in all directions.

3.7.3.2.2 Sides of Head: There shall be not more than three indications per head. The length of each indication may be the full height of the surface but no indication shall break over either edge to a depth greater than 0.031 inch or the equivalent of the 2H/3 thread depth (see Table 1), whichever is less.

3.7.3.2.3 Shank or Stem: There shall be not more than five indications. The length of any indication may be the full length of the surface but the total length of all indications shall not exceed twice the length of the surface. No indication shall break into a fillet or over an edge.

3.7.3.2.4 Threads: There shall be no indications, except as permitted in 3.7.2.6.

3.7.3.2.5 Top of Head and End of Stem: The number of indications is not restricted but the depth of any individual indication shall not exceed 0.010 inch as shown by sectioning representative samples. No indication, except those of 3.7.3.2.2, shall break over an edge.

4. QUALITY ASSURANCE PROVISIONS:

4.1 Responsibility for Inspection:

The vendor of parts shall supply all samples and shall be responsible for performing all required tests. Purchaser reserves the right to perform such confirmatory testing as deemed necessary to ensure that the parts conform to the requirements of this specification.

SAE AS7468 Revision A**4.2 Responsibility for Compliance:**

The manufacturer's system for parts production shall be based on preventing product defects, rather than detecting the defects at final inspection and then requiring corrective action to be invoked. An effective manufacturing in-process control system shall be established, subject to the approval of the purchaser, and used during production of parts.

4.3 Production Acceptance Tests:

The purpose of production acceptance tests is to check, as simply as possible, using a method which is inexpensive and representative of the part usage, with the uncertainty inherent in random sampling, that the parts comprising a production inspection lot satisfy the requirements of this specification.

4.4 Classification of Tests:

- a. Acceptance tests which are to be performed on each production inspection lot. A summary of acceptance tests is specified in Table 3.

4.5 Acceptance Test Sampling:

4.5.1 Material: Sampling for material composition on each heat shall be in accordance with AMS 5844.

4.5.2 Nondestructive Tests - Visual and Dimensional: A random sample of parts shall be taken from each production inspection lot; the size of the sample to be as specified in Table 4. The classification of dimensional characteristics shall be as specified in Table 5. All dimensional characteristics are considered defective when out of tolerance.

4.5.3 Fluorescent Penetrant Inspection: A random sample shall be selected from each production inspection lot; the size of the sample shall be as specified in Table 4 and classified as in Table 5. The sample units may be selected from those that have been subjected to and passed the visual and dimensional inspection, with additional units selected at random from the production inspection lot as necessary.

4.5.4 Macroscopic Examination: A random sample of one part shall be selected from each production inspection lot.

4.5.5 Destructive Tests: A random sample shall be selected from each production inspection lot; the size of the sample shall be as specified in Table 6. The sample units may be selected from those that have been subjected to and passed the nondestructive tests and the fluorescent penetrant inspection with additional units selected at random from the production inspection lot as necessary.

4.5.6 Acceptance Quality: Of random samples tested, acceptance quality shall be based on zero defectives.

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4.5.7 Test Specimens: Specimens for tensile testing of machined test specimens shall be of standard proportions in accordance with ASTM E 8. Specimens shall be machined from finished parts or coupons of the same lot of alloy and be processed together with the parts they represent. Specimens shall be machined from the center of parts.

4.6 Reports:

The vendor of parts shall furnish with each shipment a report stating that the chemical composition of the parts conforms to the applicable material specification, showing the results of tests to determine conformance to the room temperature ultimate tensile property, hardness, ultimate shear and fatigue properties where applicable, and stating that the parts conform to the other technical requirements. This report shall include the purchase order number, AS7468 and Revision Letter, lot number, contractor or other direct supplier of material, part number, nominal size, and quantity.

4.7 Rejected Lots:

If a production inspection lot is rejected, the vendor of parts shall perform corrective action to screen out or rework the defective parts, resubmit for acceptance tests inspection as in Table 3, or scrap the entire lot. Resubmitted lots shall be clearly identified as reinspected lots.

5. PREPARATION FOR DELIVERY:

5.1 Packaging and Identification:

5.1.1 Packaging shall be in accordance with ASTM D 3951.

5.1.2 Parts having different part numbers shall be packed in separate containers.

5.1.3 Each container of parts shall be marked to show not less than the following information:

BOLTS, UNS R30035, 260 ksi
AS7468 (or AS7468-1, as applicable)
PART NUMBER
LOT NUMBER
PURCHASE ORDER NUMBER
QUANTITY
MANUFACTURER'S IDENTIFICATION

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5.1.4 Threaded fasteners shall be protected from abrasion and chafing during handling, transportation, and storage.

6. ACKNOWLEDGMENT:

A vendor shall mention this specification number in all quotations and when acknowledging purchase orders.

7. REJECTIONS:

Parts not conforming to this specification, or to modifications authorized by purchaser, will be subject to rejection.

8. NOTES:

8.1 Hardness conversion tables for metals are presented in ASTM E 140.

8.2 Key Words:

Bolts, Procurement Specification

8.3 The change bar (I) located in the left margin is for the convenience of the user in locating areas where technical revisions, not editorial changes, have been made to the previous issue of this document.

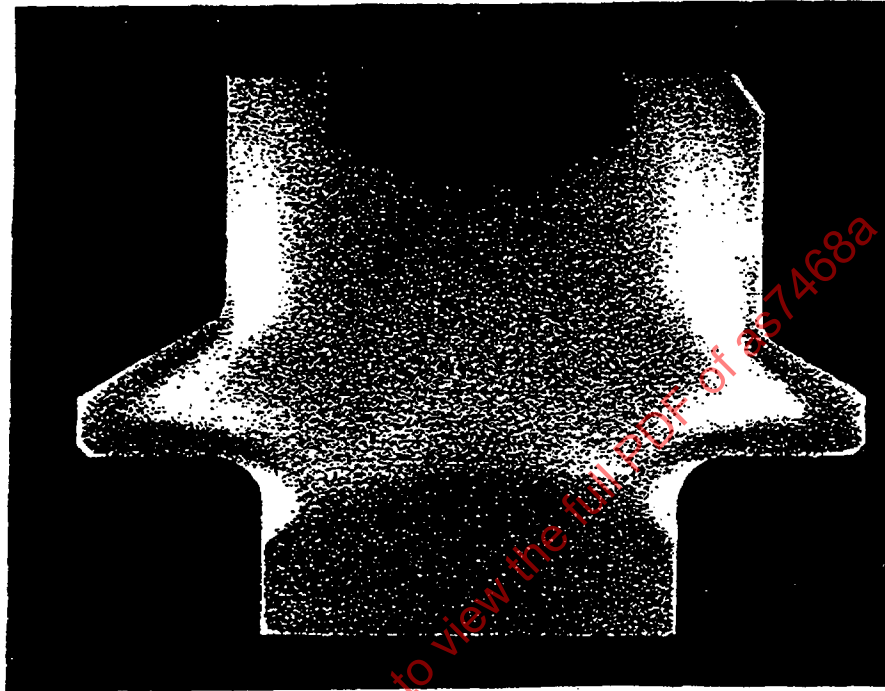
PREPARED BY SAE COMMITTEE E-25,
GENERAL STANDARDS FOR AEROSPACE PROPULSION SYSTEMS

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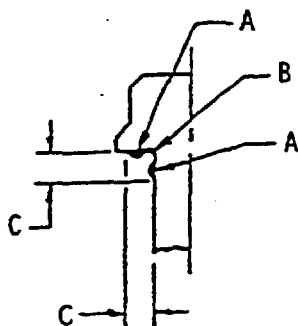
**FIGURE 1 - Satisfactory Grain Flow, Headed
Blank, Before Heat Treatment**

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**FIGURE 1A - Satisfactory Heat Pattern
Headed Blank, Before Heat Treatment**

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Nominal Bolt Diameter, inch	C max inch
Up to 0.3125, excl	0.062
0.3125 and 0.375	0.094
0.4375 to 0.625, incl	0.125
0.750 to 1.000, incl	0.156
Over 1.000	0.188

FIGURE 2 - Permissible Distortion From Fillet Working

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FIGURE 3 - Flow Lines, Rolled Thread

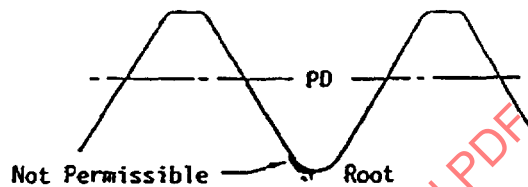


FIGURE 4 - Root Defects, Rolled Thread

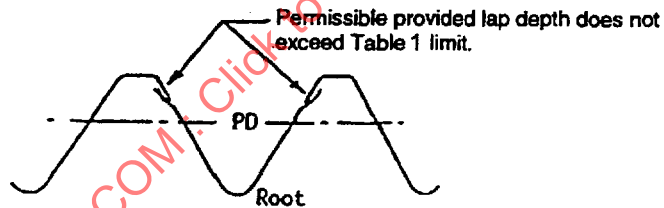


FIGURE 5 - Laps Above Pitch Diameter Extending Towards Crest, Rolled Thread

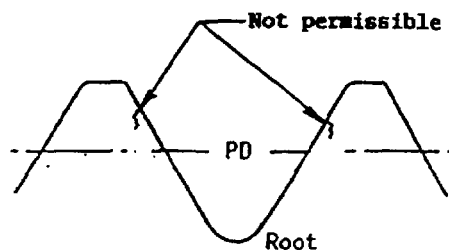


FIGURE 6 - Laps Above PD Extending Toward Root, Rolled Thread

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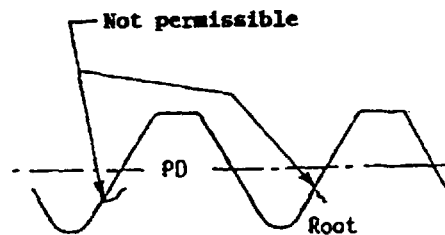
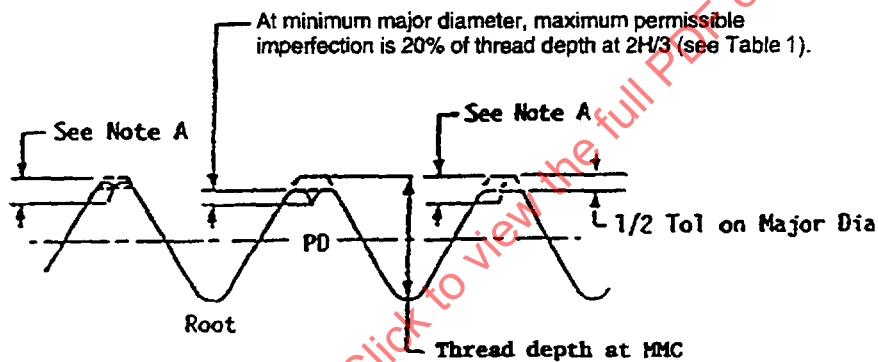


FIGURE 7 - Laps Below PD Extending in Any Direction, Rolled Thread



Note A: Maximum depth of imperfection equals 20% of thread depth at $2H/3$ plus $1/2$ the difference of the actual major diameter and minimum major diameter.

FIGURE 8 - Crest Craters and Crest Laps, Rolled Thread

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TABLE 1 - UNJ Ext Thread Depth at 2H/3 and Allowable Thread Lap Depth

Thread Pitches Per Inch n	UNJ Ext Thread Depth at 2H/3 inch	Allowable Thread Lap Depth inch
40	0.0144	0.0029
36	0.0160	0.0032
32	0.0180	0.0036
28	0.0206	0.0041
24	0.0241	0.0048
20	0.0289	0.0058
18	0.0321	0.0064
16	0.0361	0.0072
14	0.0412	0.0082
13	0.0444	0.0089
12	0.0481	0.0096
11	0.0525	0.0105
10	0.0577	0.0115
9	0.0642	0.0128
8	0.0722	0.0144

Note 1: Allowable lap depth is based upon 20% of UNJ external thread depth at 2H/3 in accordance with MIL-S-8879, and is calculated as follows:

$$\text{Ext thd depth} = 2H/3 = (2/3) (\cos 30^\circ)/n = 0.57735/n$$

$$\text{Lap depth} = 0.2(2H/3) = 0.2(2/3)(\cos 30^\circ)/n = 0.11547/n$$

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TABLE 2 - Test Loads

Thread Size UNJF-3A	Ultimate Tensile Strength Test Load, lbf min Room Temp.	Fatigue Test Load lbf max Room Temp.	Ultimate Double Shear Test Load lbf min Room Temp.
0.190 -32	5650	2930	8220
0.250 -28	10100	5240	14200
0.3125-24	16000	8300	22200
0.375 -24	24700	12800	32000
0.4375-20	33500	17400	43600
0.500 -20	44600	23200	56900
0.5625-18	56600	29400	72100
0.625 -18	70800	36800	89000
0.750 -16	102800	53400	128000
0.875 -14	140200	72800	174000
1.000 -12	182700	94900	228000

Bolt - Double Hexagon Head and Spline Drive Head as in MS21296

NOTE 1: Requirements above apply to parts with UNJF thread form in accordance with MIL-S-8879, Class 3A tolerances. For nominal thread major diameter 0.3125 inch and smaller, the diameter of the area on which stress is based for ultimate tensile and fatigue test loads is 98% of the maximum pitch diameter for UNJ thread, calculated from Equation 1:

$$A_1 = .7864[.98(d - (.6495/n))]^2 \quad (\text{Eq. 1})$$

where, A_1 = area at 98% of maximum PD
 d = maximum major diameter
 n = number of thread pitches per inch

For nominal thread major diameter larger than 0.3125 inch, the diameter of the area on which stress for ultimate tensile and fatigue test loads is based is the maximum pitch diameter, calculated from Equation 2:

$$A_2 = .7854[d - (.6495/n)]^2 \quad (\text{Eq. 2})$$

The area upon which stress for ultimate double shear strength is based is twice the area at the nominal shank diameter which is equal to the basic major diameter of the thread and derived from Equation 3:

$$A_3 = .7854(d)^2 \quad (\text{Eq. 3})$$

where, A_3 = area of the nominal close tolerance, full shank diameter
 d = nom shank dia = basic major dia of ext thread